

Massachusetts Institute of Technology
DEPARTMENT OF MECHANICAL ENGINEERING
Center for Ocean Engineering

2.611 SHIP POWER and PROPULSION

Problem Set #4 Basic Thermodynamic Cycles, Due: November 2, 2006

1. First Law refresher-

- a) Write the generic first law thermodynamic equation for a single inlet and single exit flow.
- b) What does this equation not take into account?
- c) Discuss the adiabatic process, the polytropic process and what it means for a process to be reversible.

2. Air at 10°C and 80kPa enters the diffuser of a jet engine steadily with a velocity of 200 m/s . The inlet area of the diffuser is 0.5 m^2 . The air leaves the diffuser with a velocity that is very small compared with the inlet velocity. (Assume the diffuser is the system, flow is steady and air is an ideal gas ($R=0.287\text{ (kPa}\cdot\text{m}^3)/(\text{kg}\cdot\text{K})$, enthalpy @ $283\text{ Kelvin} \sim 283\text{ kJ/kg}$)

Determine:

- a) The mass flow rate of the air
- b) The temperature of the air leaving the diffuser.



3. Consider the tank system below. Tank A has a volume of 100ft^3 and initially contains R134a at a pressure of 100 kPa and a temperature of 313 Kelvin . The compressor evacuates tank A and charges tank B. Tank B is initially evacuated and is of such volume that the final pressure of the R134a in tank B is 800 kPa . Temperature remains constant. Determine the work done by the compressor.

4. Steam enters an adiabatic turbine at 8 MPa and 500°C with a mass flow rate of 3 kg/s and leaves at 30 kPa . The isentropic efficiency of the turbine is $.9$. Neglecting kinetic energy, determine:

- a) Temperature at the turbine exit
- b) Power output

5. A marine steam plant operates as a simple Rankine cycle with a turbine inlet temperature and pressure of $600\text{ }^{\circ}\text{C}$ and 4 MPa . The condenser operates at a pressure of 20 kPa . Assume that the turbine isentropic efficiency is 80% and the pump isentropic efficiency is 90% .

- a. Sketch the cycle on T-s and h-s diagrams.
- b. Determine the steam quality at exit from the turbine.
- c. Determine the specific enthalpy change across each component.
- d. Determine the net power of the cycle with a mass flow rate of 3 kg/s .
- e. Determine the thermal efficiency of the Rankine cycle.