

# Today's goals

- **Last week**
  - Frequency response= $G(j\omega)$
  - Bode plots
- **Today**
  - Using Bode plots to determine stability
    - Gain margin
    - Phase margin
  - Using frequency response to determine transient characteristics
    - damping ratio / percent overshoot
    - bandwidth / response speed
    - steady-state error
  - Gain adjustment in the frequency domain

# Gain and phase margins

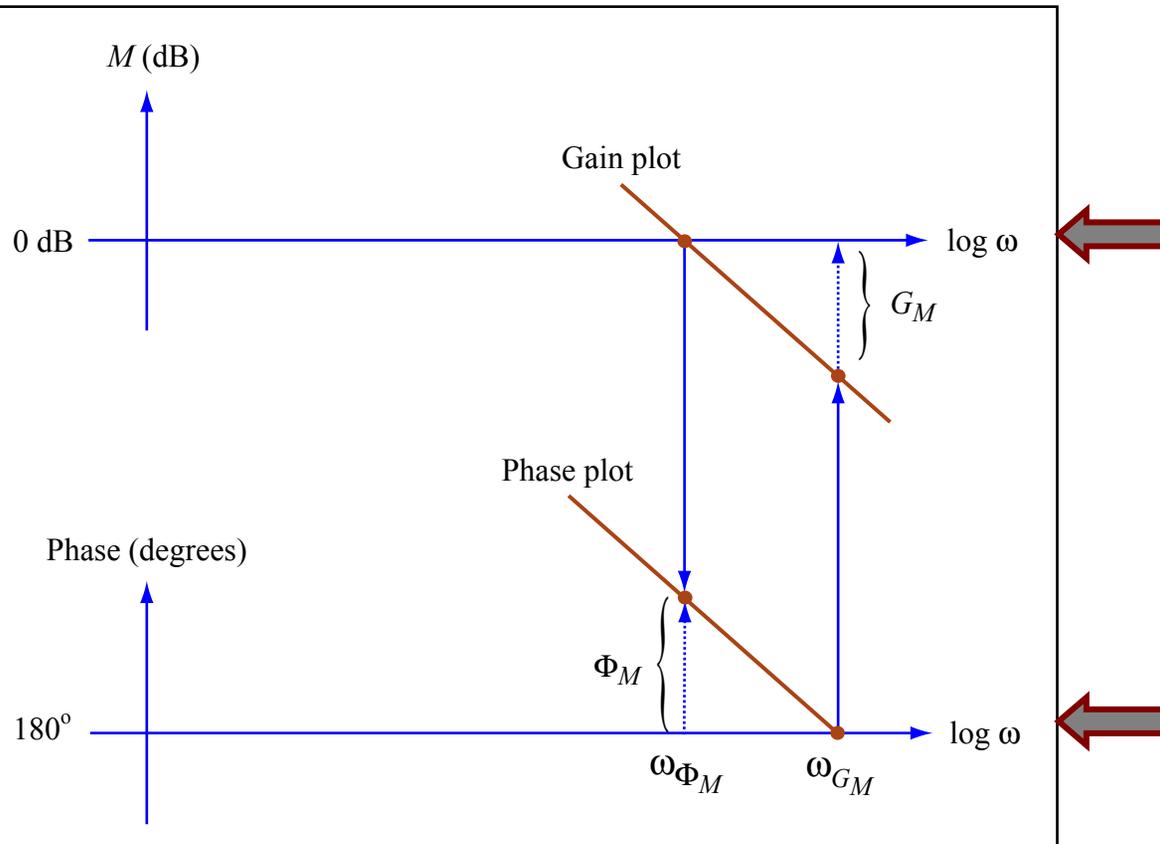


Figure by MIT OpenCourseWare.

A system is stable if the gain and phase margins are both positive

Figure 10.37

Gain margin:  
the difference (in dB) between 0dB and the system gain, computed at the frequency where the phase is  $180^\circ$

Phase margin:  
the difference (in  $^\circ$ ) between the system phase and  $180^\circ$ , computed at the frequency where the gain is 1 (i.e., 0dB)

# Example 1

Open-Loop Transfer Function

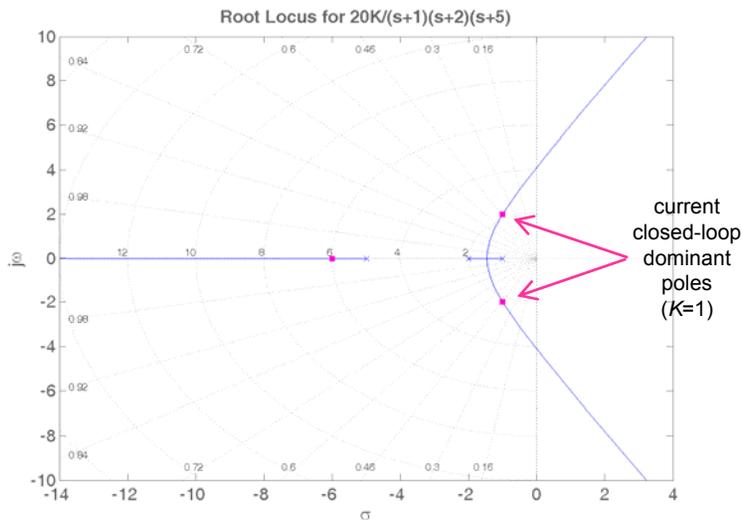
$$KG(s)H(s) = \frac{20K}{(s+1)(s+2)(s+5)}$$

DC gain = 20 =  $20\log_{10}20(\text{dB}) \approx 6\text{dB}$

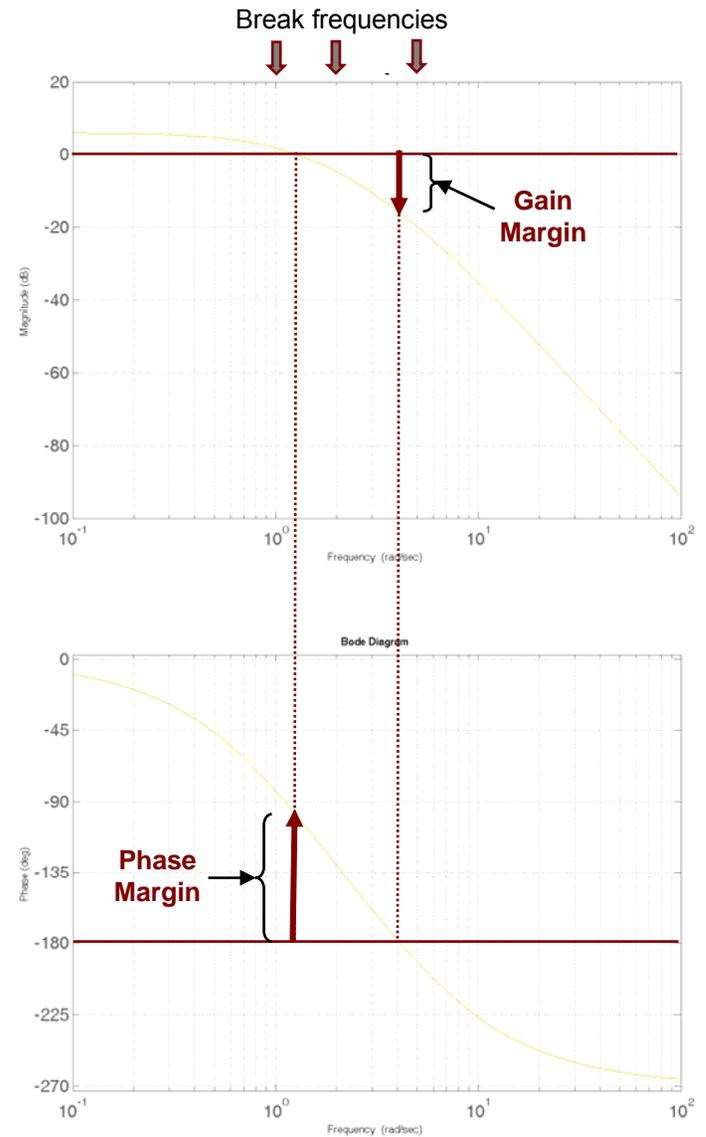
Break (cut-off) frequencies: 1, 2, 5 rad/sec.

Final gain slope:  $-60\text{ dB/dec.}$

Total phase change:  $-270^\circ$ .



Increasing the closed-loop gain by an amount equal to the G.M. (i.e., setting  $K=+\text{G.M. dB}$  or more) will destabilize the system

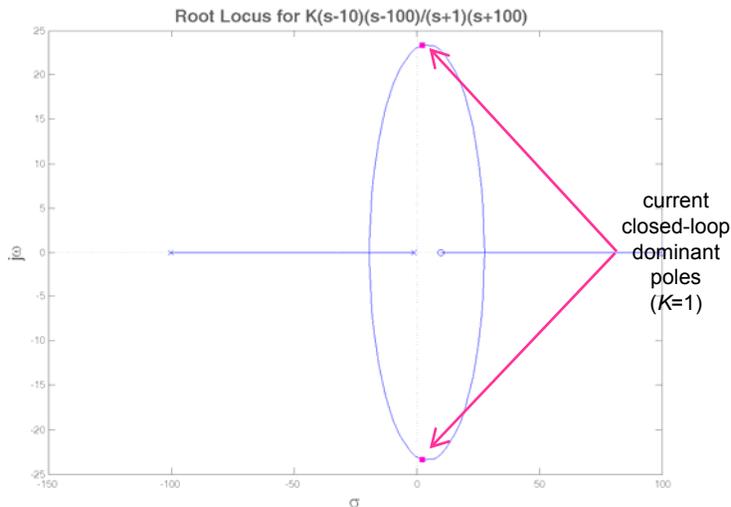
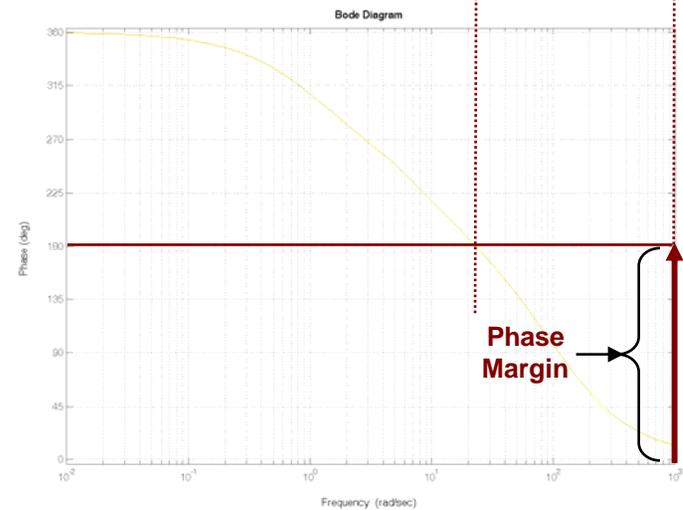
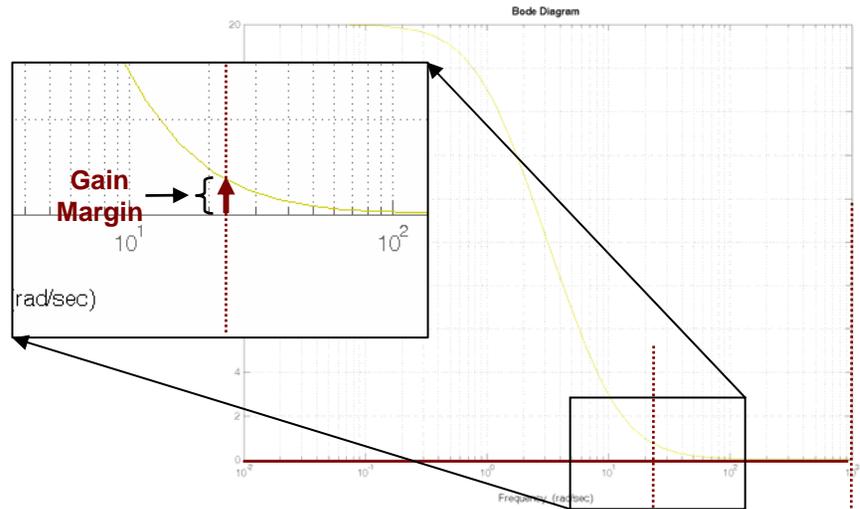


# Example 2

Open-Loop Transfer Function

$$KG(s)H(s) = \frac{K(s - 10)(s - 100)}{(s + 1)(s + 100)}$$

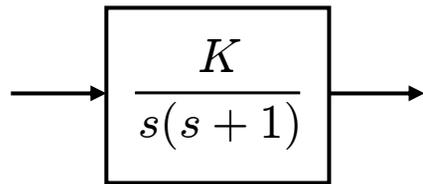
negative Gain Margin  
 → **closed-loop system is unstable**



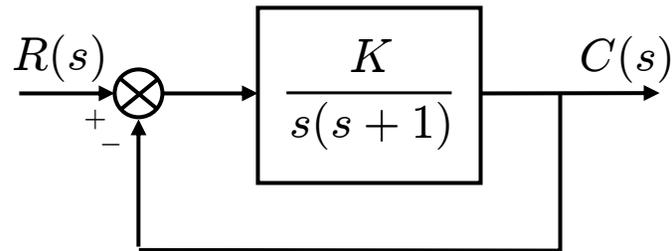
Reducing the closed-loop gain by an amount equal to the G.M.  
 (i.e., setting  $K = -\text{G.M. dB}$  or less) will stabilize the system

# Transient from *closed-loop* frequency response /1

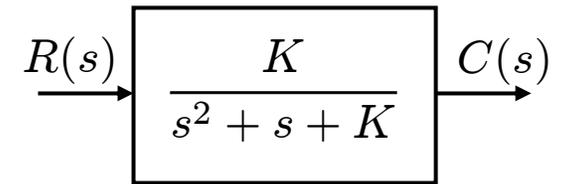
Consider a 1st-order system with ideal integral control:



Open-Loop system

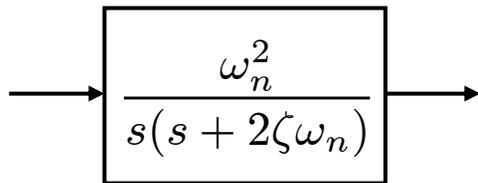


Closed-Loop system

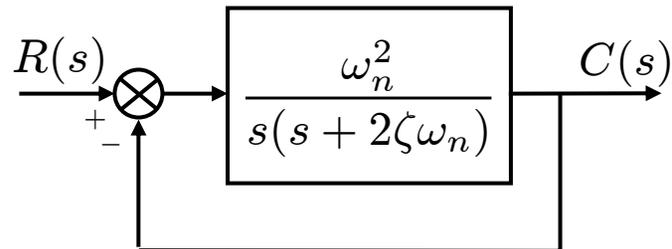


Equivalent block diagram  
for the Closed-Loop system

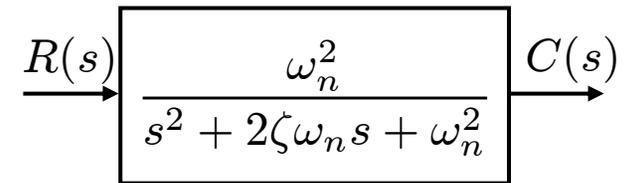
More generally, with the definition  $K \equiv \omega_n^2$ :



Open-Loop system



Closed-Loop system



Equivalent block diagram  
for the Closed-Loop system

# Transient from *closed-loop* frequency response /2

Closed-Loop Transfer Function:  $\frac{C(s)}{R(s)} \equiv T(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$

Frequency Response magnitude:  $M(\omega) = |T(s)| = \frac{\omega_n^2}{\left\{(\omega_n^2 - \omega^2)^2 + 4\zeta^2\omega_n^2\omega^2\right\}^{1/2}}$

Frequency response magnitude peaks at frequency  $\omega_p = \omega_n \sqrt{1 - 2\zeta^2}$ .

Frequency response peak magnitude is  $M_p = \frac{1}{2\zeta\sqrt{1 - \zeta^2}}$ .

## Bandwidth:

The frequency where the magnitude drops by 3dB below the DC magnitude

$$\omega_{\text{BW}} = \sqrt{(1 - 2\zeta^2) + \sqrt{4\zeta^4 - 4\zeta^2 + 2}}$$

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Please see: Fig. 10.39 in Nise, Norman S. *Control Systems Engineering*. 4th ed. Hoboken, NJ: John Wiley, 2004.

# Transient from *closed-loop* frequency response /3

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Please see: Fig. 10.40 and 10.41 in Nise, Norman S. *Control Systems Engineering*. 4th ed. Hoboken, NJ: John Wiley, 2004.

# Transient from *open-loop* phase diagrams

Relationship between phase margin  $\Phi_M$   
and damping ratio:

$$\Phi_M = \tan^{-1} \frac{2\zeta}{\sqrt{-2\zeta^2 + \sqrt{1 + 4\zeta^2}}}$$



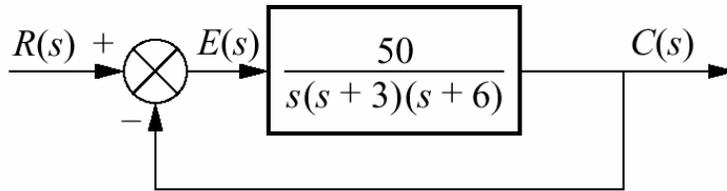
Open-Loop gain  
*vs* Open-Loop phase  
at frequency  $\omega = \omega_{BW}$   
(*i.e.*, when Closed-Loop gain  
is 3dB below the Closed-Loop DC gain.)



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Please see: Fig. 10.48 and 10.49 in Nise, Norman S. *Control Systems Engineering*. 4th ed. Hoboken, NJ: John Wiley, 2004.

# Example



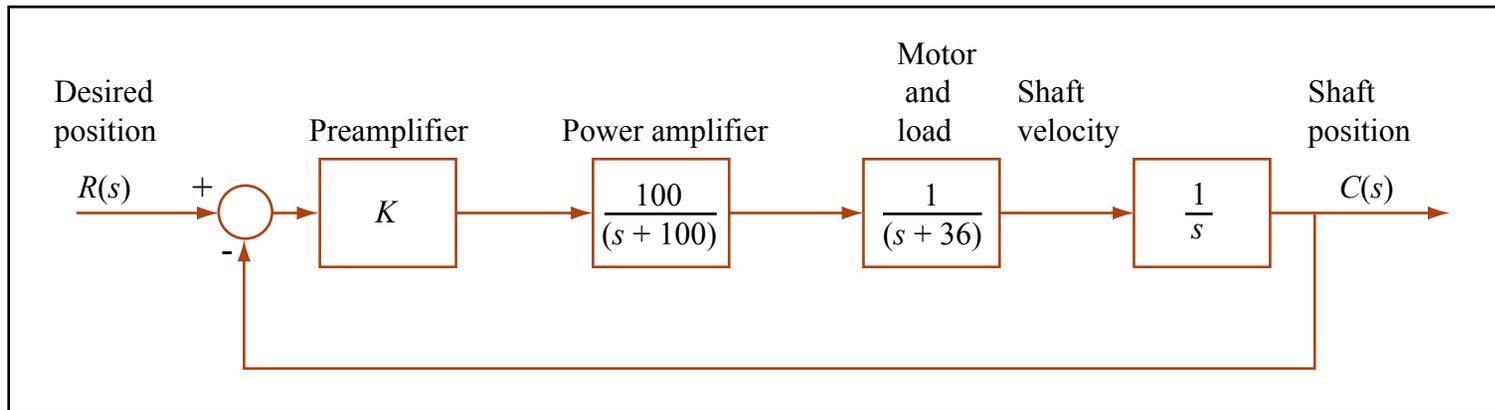
**Bandwidth from frequency response:**  
find where  $M = -6 \sim -7.5\text{dB}$  while  $\Phi = -135^\circ \sim -225^\circ$   
 $\Rightarrow \omega_{\text{BW}} \approx 3.5\text{rad/sec.}$

**Damping ratio from phase margin:**  
Find phase margin ( $\approx 35^\circ$ )  
and substitute into plot ( $\zeta \approx 0.32$ ).

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Please see: Fig. 10.50 and 10.48 in Nise, Norman S. *Control Systems Engineering*. 4th ed. Hoboken, NJ: John Wiley, 2004.

# Example: Proportional control in the frequency domain



Figures by MIT OpenCourseWare.

**Specification:** 9.5% overshoot.

For 9.5% overshoot, the required damping ratio is  $\zeta = 0.6$ .  
Using the damping ratio–phase margin relationship, we find

$$\Phi_M = \tan^{-1} \frac{2\zeta}{\sqrt{-2\zeta^2 + \sqrt{1 + 4\zeta^4}}} \Rightarrow \Phi_M = 59.2^\circ.$$

Before compensation, the phase margin was  $\approx 85^\circ$   
(see the Bode plot on the right.)

We must reduce the phase margin to  $59.2^\circ$ ,

*i.e.* the Bode magnitude must be 0dB

when the Bode phase is  $-180^\circ + 59.2^\circ = -120.8^\circ$ .

This occurs when  $\omega \approx 15^\circ$  and

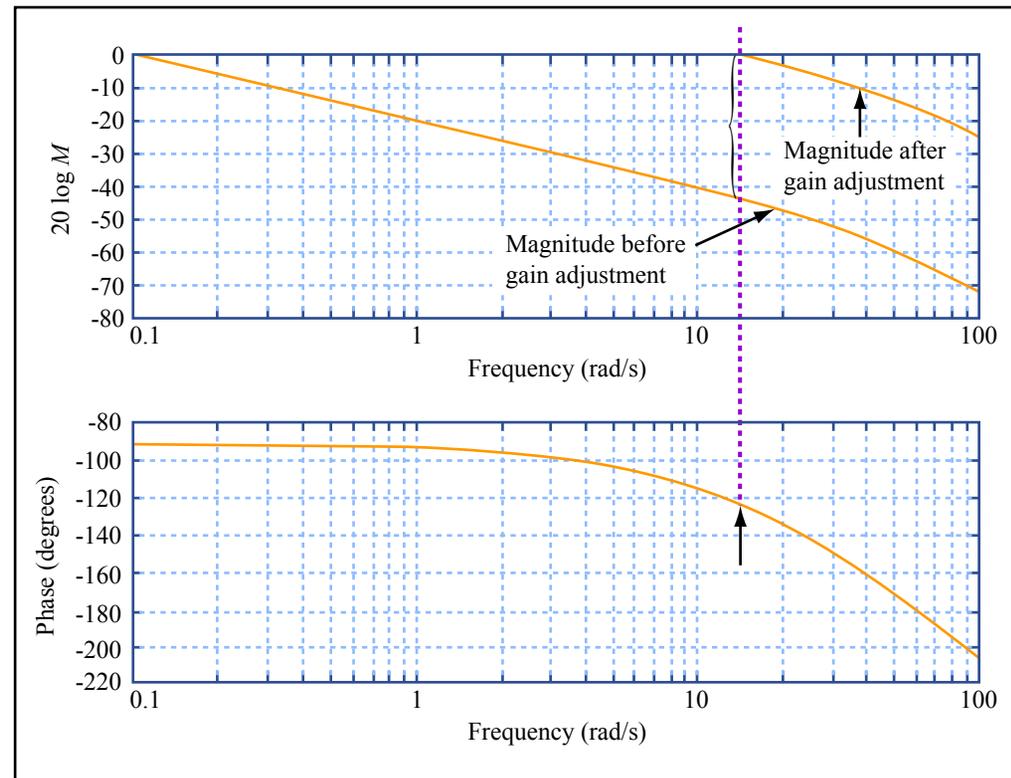
we can see that the required gain adjustment is  $\approx 44$ dB.

What is the total gain for the compensator?

In our uncompensated Bode plot,  $M = 1$  when  $\omega = 0.1 \Rightarrow$   
the uncompensated gain is  $K \approx 3.6$ .

After compensation, the gain (in dB) should be

$$20\log 3.6 + 44 \approx 11 + 44 = 55 \Rightarrow K \approx 570.$$



# Gain adjustment for phase margin specification

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Please see: Fig. 11.1 in Nise, Norman S. *Control Systems Engineering*. 4th ed. Hoboken, NJ: John Wiley, 2004.

# Steady-state errors from the frequency response

Type 0 system (no free integrators)    Type 1 system (one free integrator)    Type 2 system (two free integrators)

$$G(s) = K \frac{\prod (s + z_k)}{\prod (s + p_k)}$$

$$G(s) = K \frac{\prod (s + z_k)}{s \prod (s + p_k)}$$

$$G(s) = K \frac{\prod (s + z_k)}{s^2 \prod (s + p_k)}$$

Steady-state **position** error

Steady-state **velocity** error

Steady-state **acceleration** error

$$e_\infty = \frac{1}{1 + K_p}, \text{ where}$$

$$K_p \equiv K \frac{\prod z_k}{\prod p_k}$$

= DC gain.

$$e_\infty = \frac{1}{K_v}, \text{ where}$$

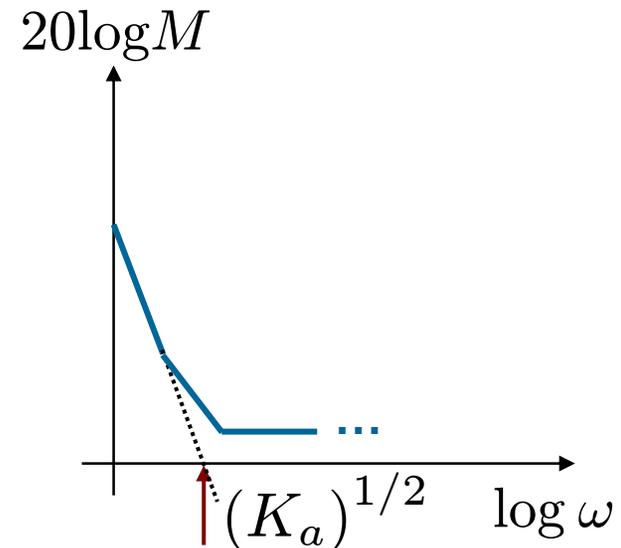
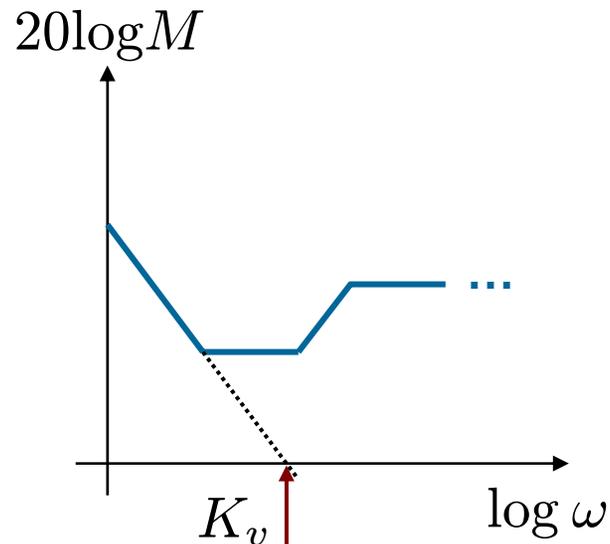
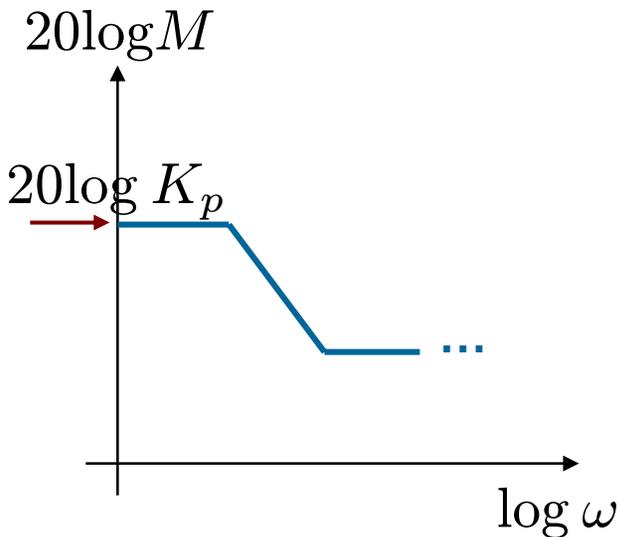
$$K_v \equiv K \frac{\prod z_k}{\prod p_k}$$

=  $\omega$ -axis intercept.

$$e_\infty = \frac{1}{K_a}, \text{ where}$$

$$K_a \equiv K \frac{\prod z_k}{\prod p_k}$$

=  $(\omega\text{-axis intercept})^2$ .



# Example

Type 0; steady-state position error

$$20\log K_p = 25 \Rightarrow e_\infty = 0.0532$$

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Please see: Fig. 10.52 in Nise, Norman S. *Control Systems Engineering*.  
4th ed. Hoboken, NJ: John Wiley, 2004.

Type 1; steady-state velocity error

$$M = 0\text{dB when } \omega = 0.55 \Rightarrow e_\infty = 1.818$$

Type 2; steady-state acceleration error

$$M = 0\text{dB when } \omega = 3 \Rightarrow e_\infty = 0.111$$