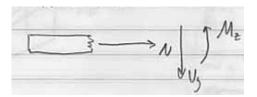
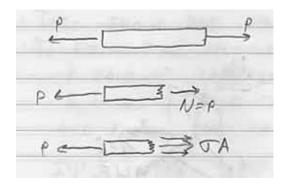
## 2.001 - MECHANICS AND MATERIALS I

## MULTI-AXIAL STRESS AND STRAIN

Recall: Internal Forces and Moments



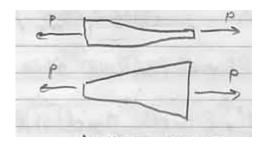
## Axial stress from uniaxial loading



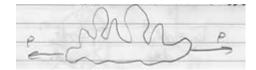
$$\sigma = \frac{P}{A}$$

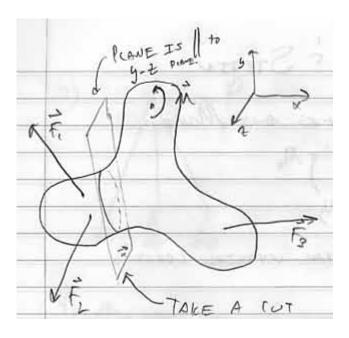
Note:  $\sigma$  is an average axial stress.

For slender (long, thin) objects A is either uniform or slowly-varying.

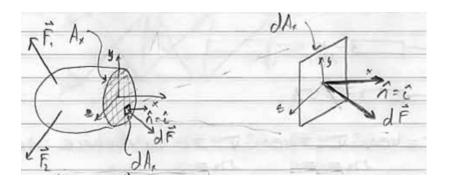


What about a more general case?





Note:  $d\vec{F}$  is not in general parallel to  $\hat{n}$ .



$$\begin{split} d\vec{F}^{(i)} &= dF_x^{(i)} \hat{i} + dFyx^{(i)} \hat{j} + dF_z^{(i)} \hat{k} \\ \hat{n} &= \hat{i} \\ N_{totalforce} &= \int_{A_x} dF_x^{(i)} \\ \bar{\sigma} &= N_{totalforce}/A_x \end{split}$$

Traction:  $\vec{t}$  Force per unit area at a point

$$\vec{t}^{(i)} = \frac{dF^{(i)}}{dA_x} = \frac{dF^{(i)}_x}{dA_x}\hat{i} + \frac{dF^{(i)}_y}{dA_x}\hat{j} + \frac{dF^{(i)}_z}{dA_x}\hat{k}$$

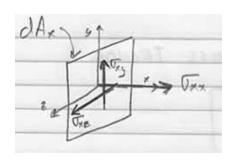
$$\sigma_x x = \frac{dF^{(i)}_x}{dA_x} \text{ (Normal Stress)}$$

$$\sigma_x y = \frac{dF^{(i)}_y}{dA_x} \text{ (Shear Stress)}$$

$$\sigma_x z = \frac{dF^{(i)}_z}{dA_x} \text{ (Shear Stress)}$$

So:

$$\vec{t}^{(i)} = \sigma_{xx}\hat{i} + \sigma_{xy}\hat{j} + \sigma_{xz}\hat{k}$$



$$N_{TOTALFACE} = \int_{A_x} \sigma_{xx} dA_x$$

$$V_{TOTALFACE_y} = \int_{A_x} \sigma_{xu} dA_x$$

$$N_{TOTALFACE_z} = \int_{A_x} \sigma_{xz} dA_x$$

To find the stresses on the

y face  $\Rightarrow$  take cut on x-z plane.

 $z \text{ face} \Rightarrow take cut on x-y plane.}$ 

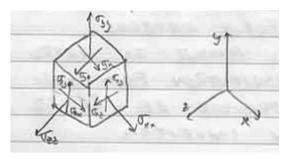
And follow the same procedure.

$$\vec{t}^{(j)} = \frac{dF_x^{(j)}}{dA_y}\hat{i} + \frac{dF_y^{(j)}}{dA_y}\hat{j} + \frac{dF_z^{(j)}}{dA_y}\hat{k}$$

$$\vec{t}^{(j)} = \sigma_{yx}\hat{i} + \sigma_{yy}\hat{j} + \sigma_{yz}\hat{k}$$

$$\begin{split} \vec{t}^{(k)} &= \frac{dF_x^{(k)}}{dA_z} \hat{i} + \frac{dF_y^{(k)}}{dA_z} \hat{j} + \frac{dF_z^{(k)}}{dA_z} \hat{k} \\ \vec{t}^{(j)} &= \sigma_{zx} \hat{i} + \sigma_{zy} \hat{j} + \sigma_{zz} \hat{k} \end{split}$$

Materal Point



See handout for more clear diagram.

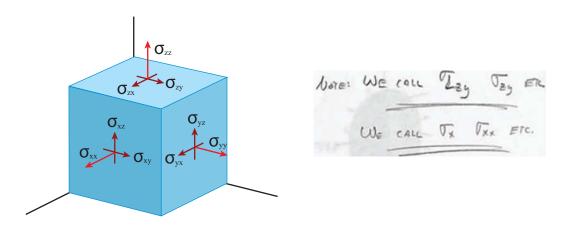


Figure by MIT OCW.

Matrix form of stress tensor

$$[\sigma ] = \left| \begin{array}{ccc} \sigma_{xx} & \sigma_{xy} & \sigma_{xz} \\ \sigma_{yx} & \sigma_{yy} & \sigma_{yz} \\ \sigma_{zx} & \sigma_{zy} & \sigma_{zz} \end{array} \right|.$$

Diagonal terms are normal stresses.

Off diagonal terms are shear stresses.