

Multidisciplinary System Design Optimization (MSDO)

Design for Value

Lecture 19

Karen Willcox

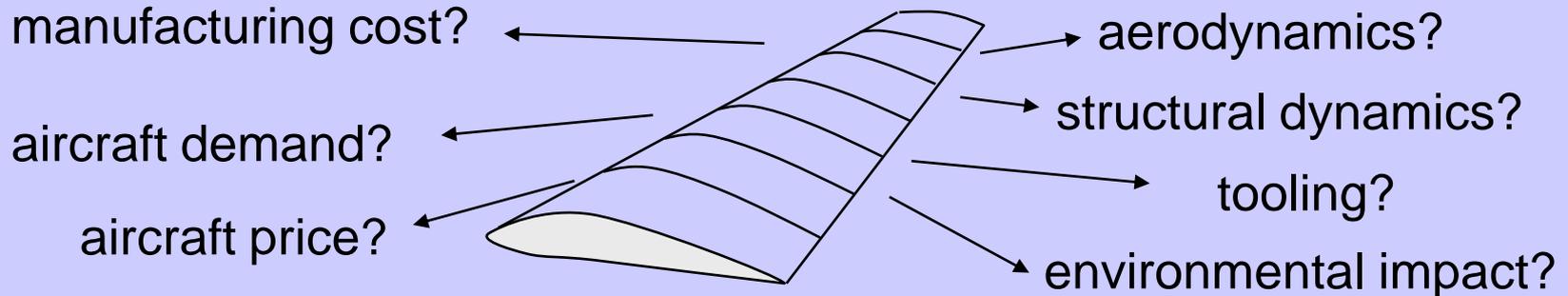
Olivier de Weck

Select figures courtesy of Jacob Markish and Ryan Peoples. Used with permission.

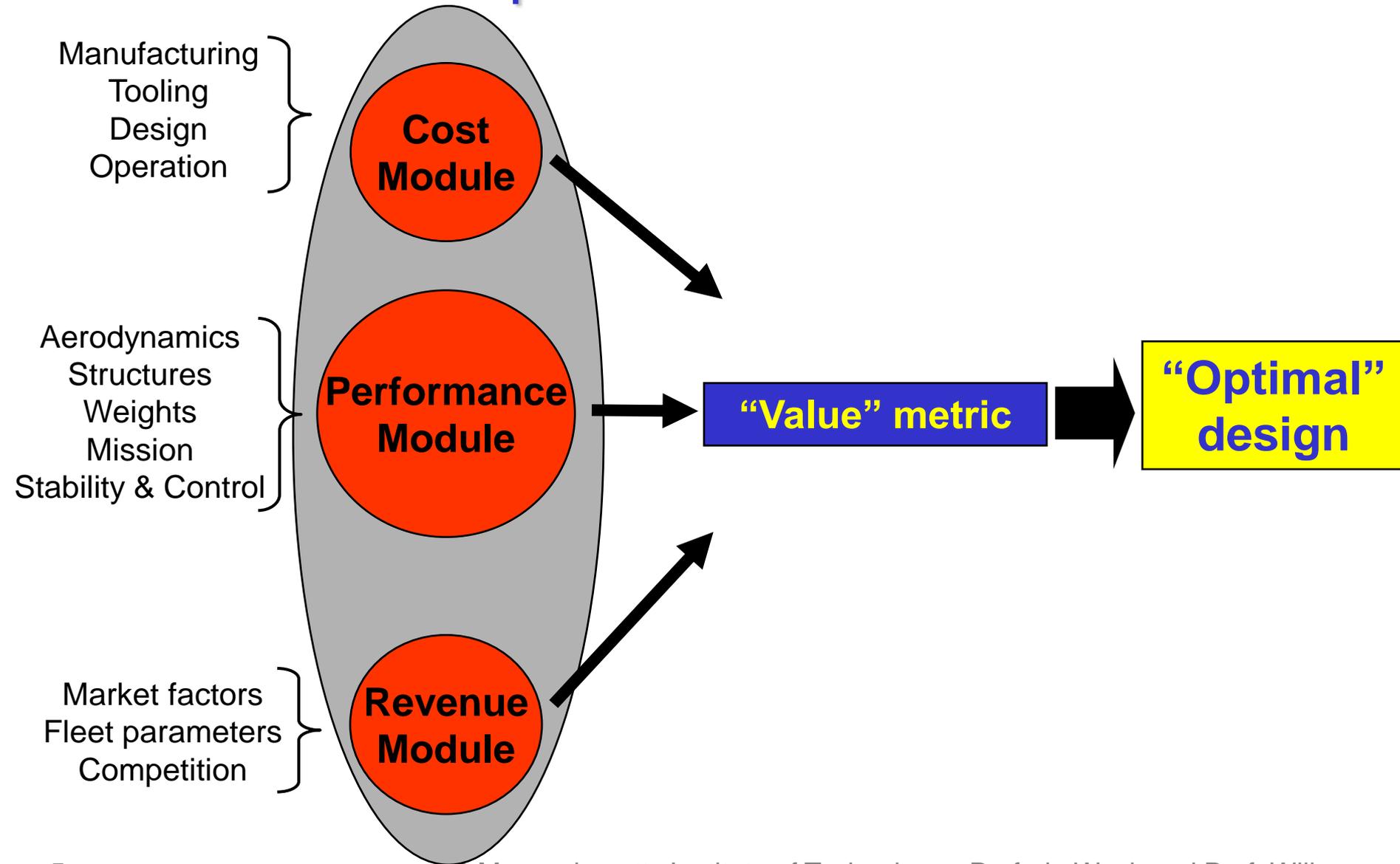
- An MDO value framework
- Lifecycle cost models
- Value metrics & valuation techniques
- Value-based MDO
- Aircraft example
- Spacecraft Example

- Traditionally, design has focused on **performance**
e.g. for aircraft design
optimal = minimum weight
- Increasingly, **cost** becomes important
- 85% of total lifecycle cost is locked in by the end of preliminary design.
- But *minimum weight \neq minimum cost \neq maximum value*
- What is an appropriate value metric?

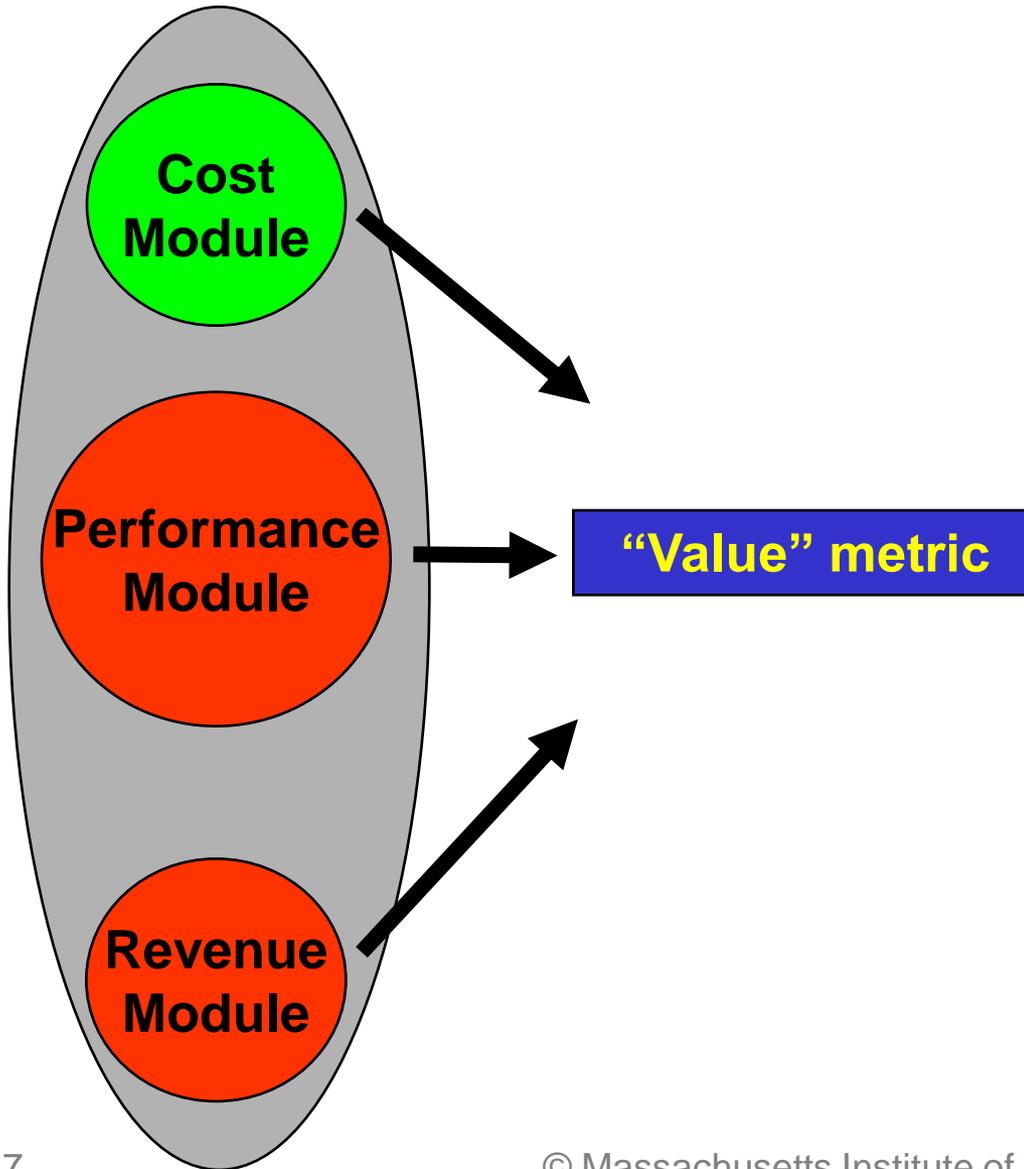
- We need to design a particular portion of the wing
- Traditional approach: balance the aero & structural requirements, minimize weight
- We should consider cost: what about an option that is very cheap to manufacture but performance is worse?



- How do we trade performance and cost?
- How much performance are we willing to give up for \$100 saved?
- What is the impact of the low-cost design on price and demand of this aircraft?
- What is the impact of this design decision on the other aircraft I build?
- What about market uncertainty?



- Cost and revenue are difficult to model
 - often models are based on empirical data
 - how to predict for new designs
- Uncertainty of market
- Long program length
- Time value of money
- Valuing flexibility
- Performance/financial groups even more uncoupled than engineering disciplines



Need to model the lifecycle cost of the system.

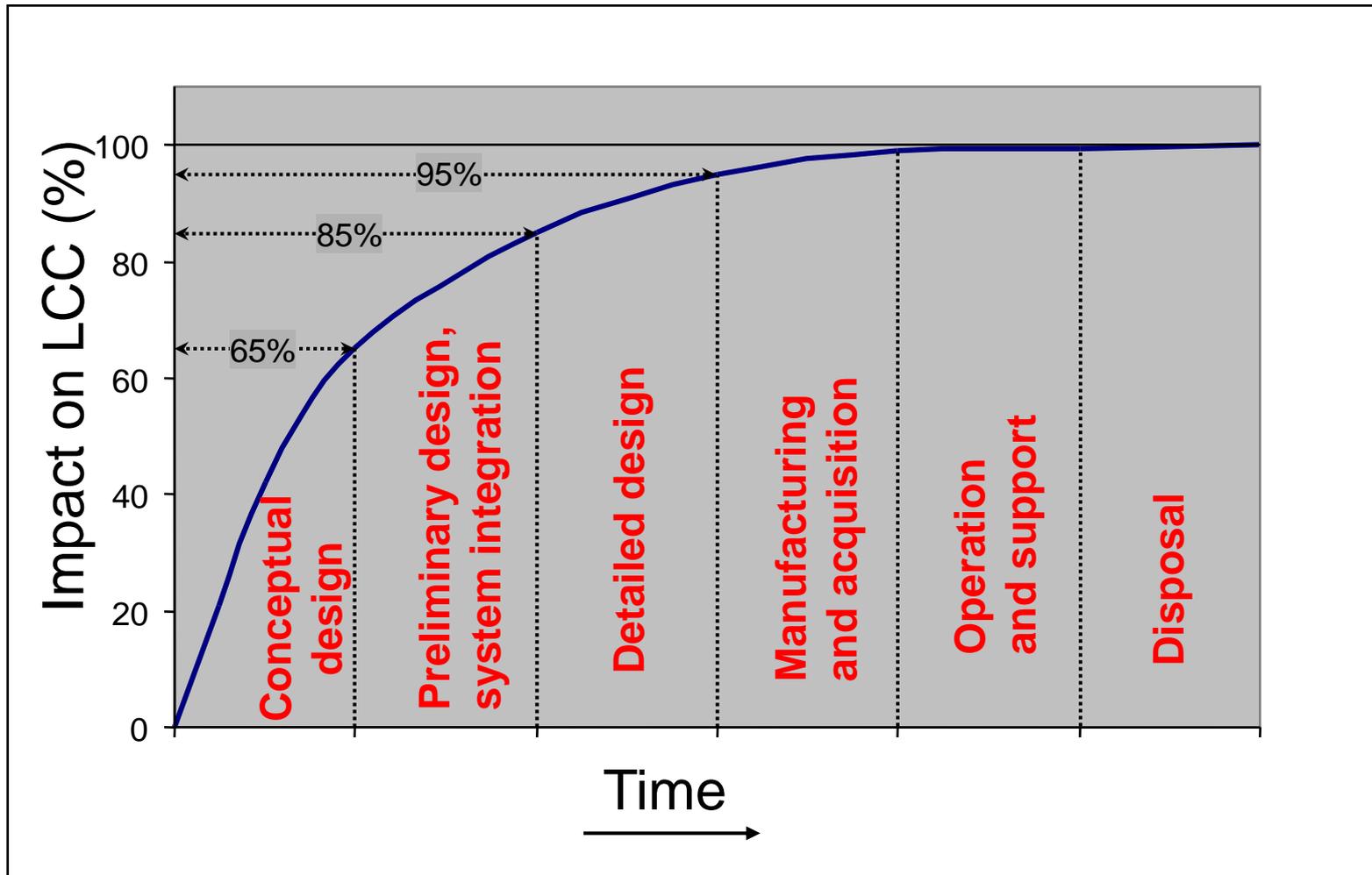
Life cycle :

Design - Manufacture -
Operation - Disposal

Lifecycle cost :

Total cost of program over
life cycle

85% of Total LCC is locked
in by the end of preliminary
design.



Cost incurred one time only:

Engineering

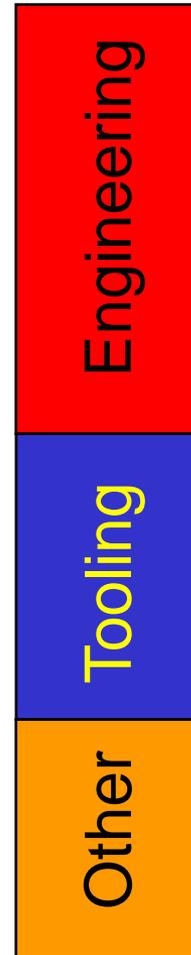
- airframe design/analysis
- configuration control
- systems engineering

Tooling

- design of tools and fixtures
- fabrication of tools and fixtures

Other

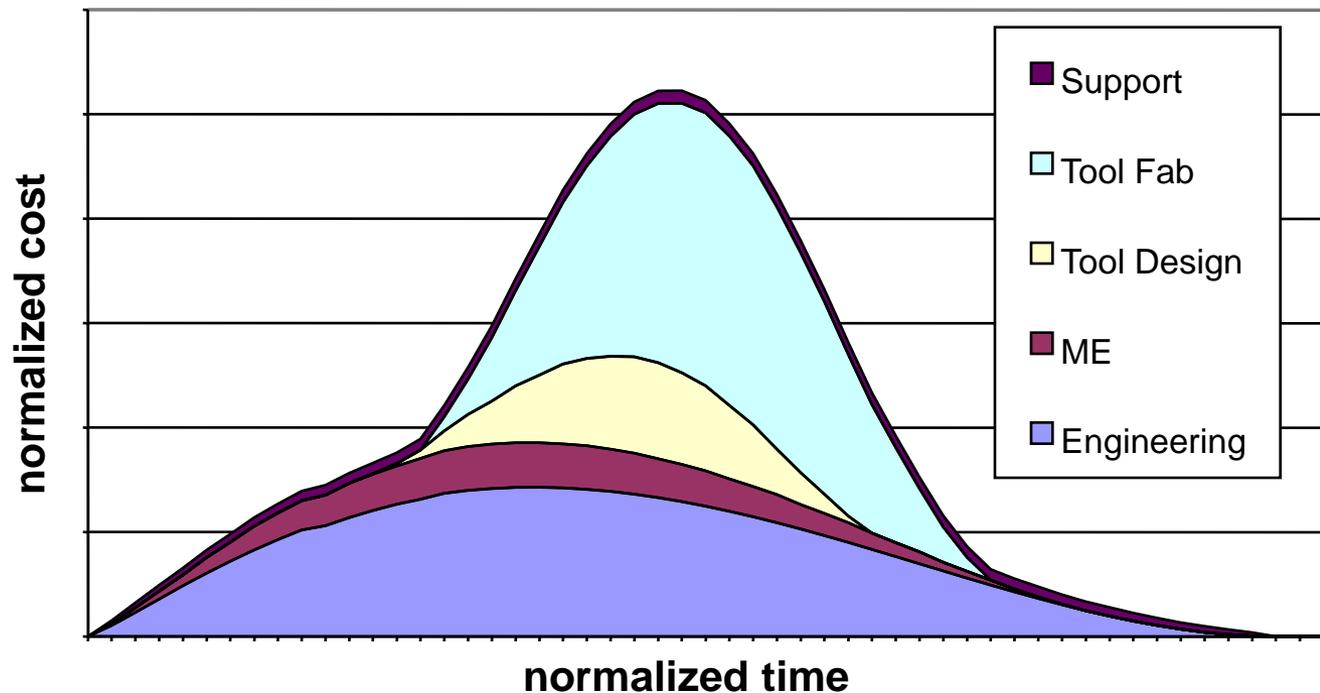
- development support
- flight testing



- Cashflow profiles based on beta curve:

$$c(t) = Kt^{\alpha-1} (1-t)^{\beta-1}$$

- Typical development time ~6 years
- Learning effects captured – span, cost



Cost incurred per unit:

Labor

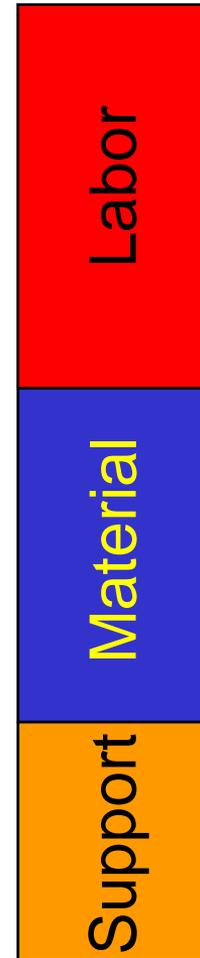
- fabrication
- assembly
- integration

Material to manufacture

- raw material
- purchased outside production
- purchased equipment

Production support

- QA
- production tooling support
- engineering support



As more units are made, the recurring cost per unit decreases.

This is the learning curve effect.

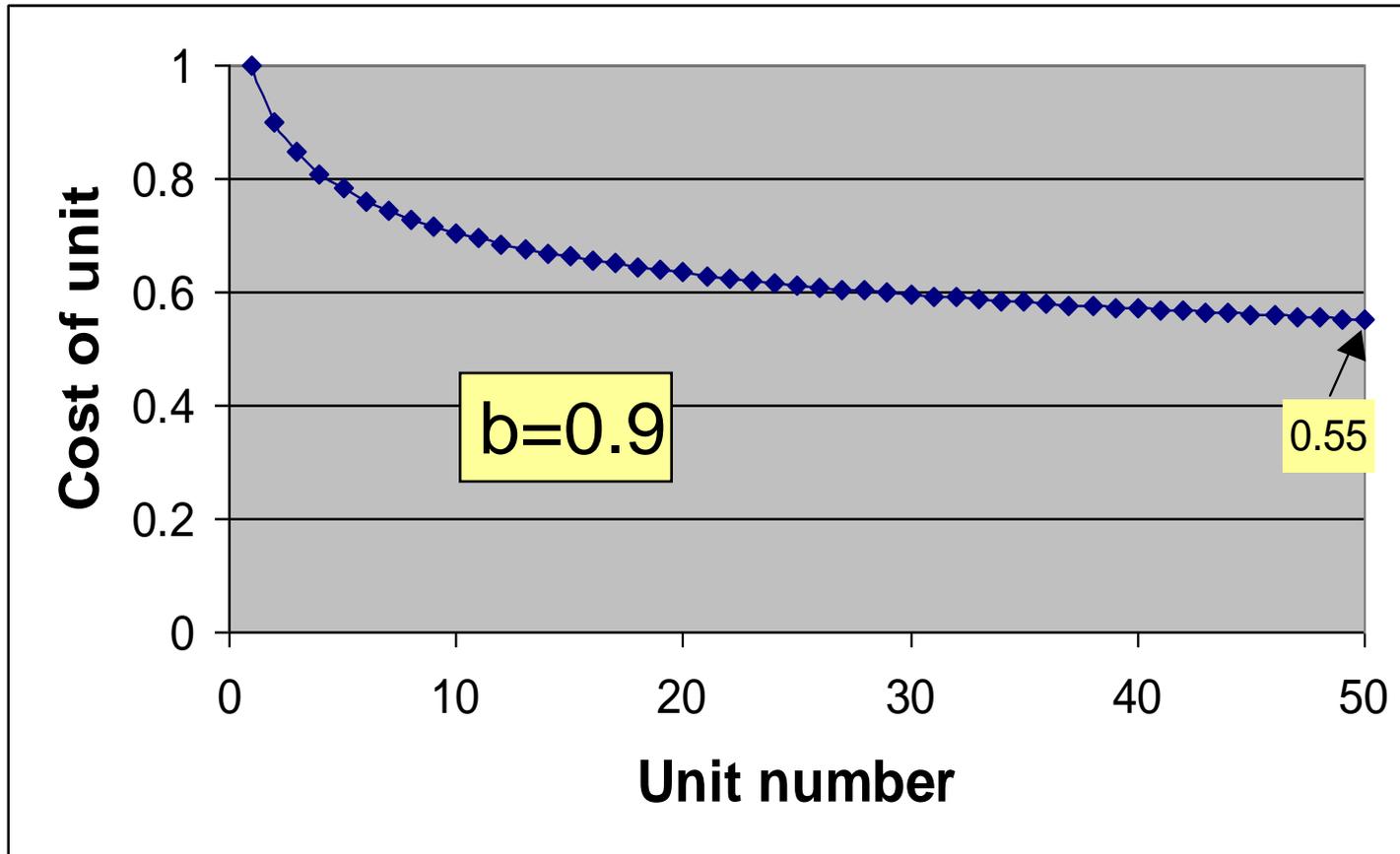
e.g. Fabrication is done more quickly, less material is wasted.

$$Y_x = Y_0 x^n$$

Y_x = number of hours to produce unit x

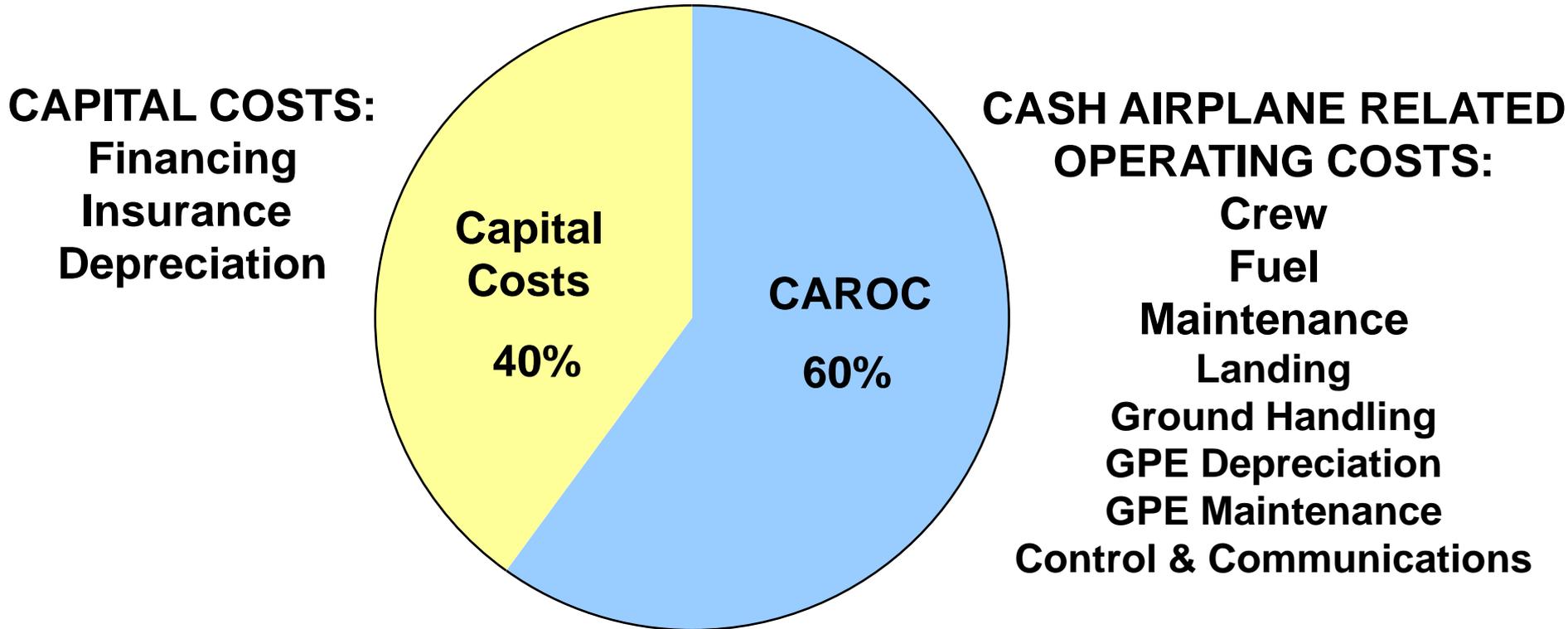
$n = \log b / \log 2$

b = learning curve factor (~80-100%)

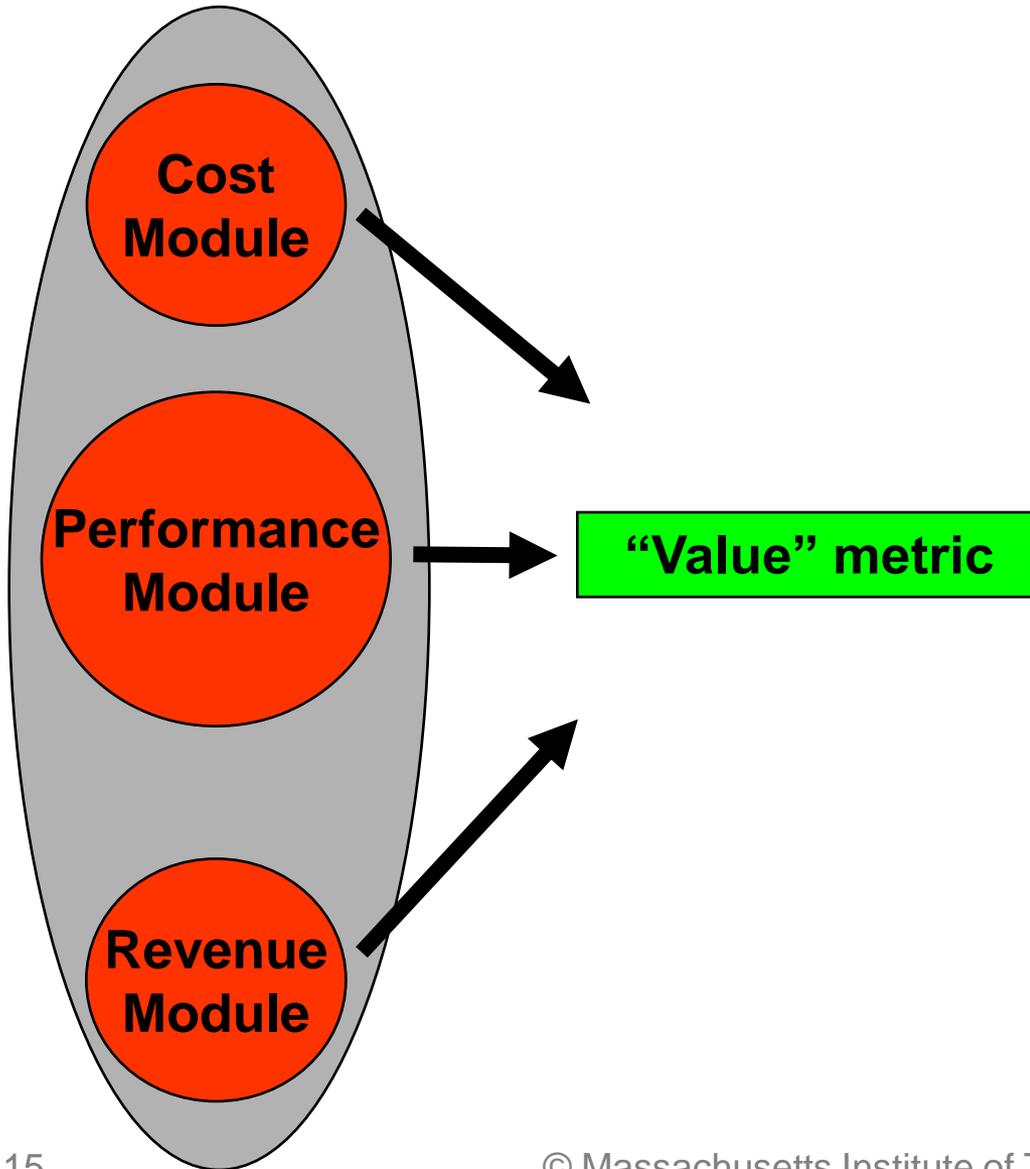


Every time production doubles, cost is reduced by a factor of 0.9

Typical LC slopes: Fab 90%, Assembly 75%, Material 98%



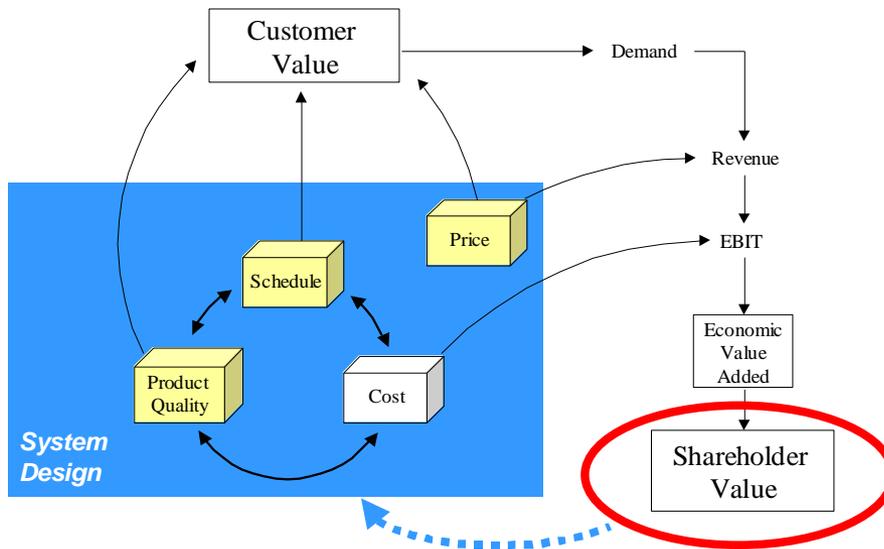
CAROC is only 60% - ownership costs are significant!



Need to provide a quantitative metric that incorporates cost, performance and revenue information.

In optimization, need to be especially carefully about what metric we choose...

- Objective function could be different for each stakeholder
e.g. manufacturer vs. airline vs. flying public
- Program related parameters vs. technical parameters
cost, price, production quantity, timing
- Traditionally program-related design uncoupled from technical design



Customer value derived from quality, timeliness, price.

Shareholder value derived from cost and revenue, which is directly related to customer satisfaction.

Traditional Metrics

performance
weight
speed

Augmented Metrics

cost
revenue
profit
quietness
emissions
commonality
...

The definition of value will vary depending on your system and your role as a stakeholder, but we must define a quantifiable metric.

Investor questions:

- How much will I need to invest?
- How much will I get back?
- When will I get my money back?
- How much is this going to cost me?
- How are you handling risk & uncertainty?

Investment Criteria

- Net present value
- Payback
- Discounted payback
- Internal rate of return
- Return on investment

- Measure of present value of various cash flows in different periods in the future
- Cash flow in any given period discounted by the value of a dollar today at that point in the future
 - “Time is money”
 - A dollar tomorrow is worth less today since if properly invested, a dollar today would be worth more tomorrow
- Rate at which future cash flows are discounted is determined by the “discount rate” or “hurdle rate”
 - Discount rate is equal to the amount of interest the investor could earn in a single time period (usually a year) if s/he were to invest in a “safer” investment

- Forecast the cash flows, C_0, C_1, \dots, C_T of the project over its economic life
 - Treat investments as negative cash flow
- Determine the appropriate opportunity cost of capital (i.e. determine the discount rate r)
- Use opportunity cost of capital to discount the future cash flow of the project
- Sum the discounted cash flows to get the net present value (NPV)

$$NPV = C_0 + \frac{C_1}{1+r} + \frac{C_2}{(1+r)^2} + \dots + \frac{C_T}{(1+r)^T}$$

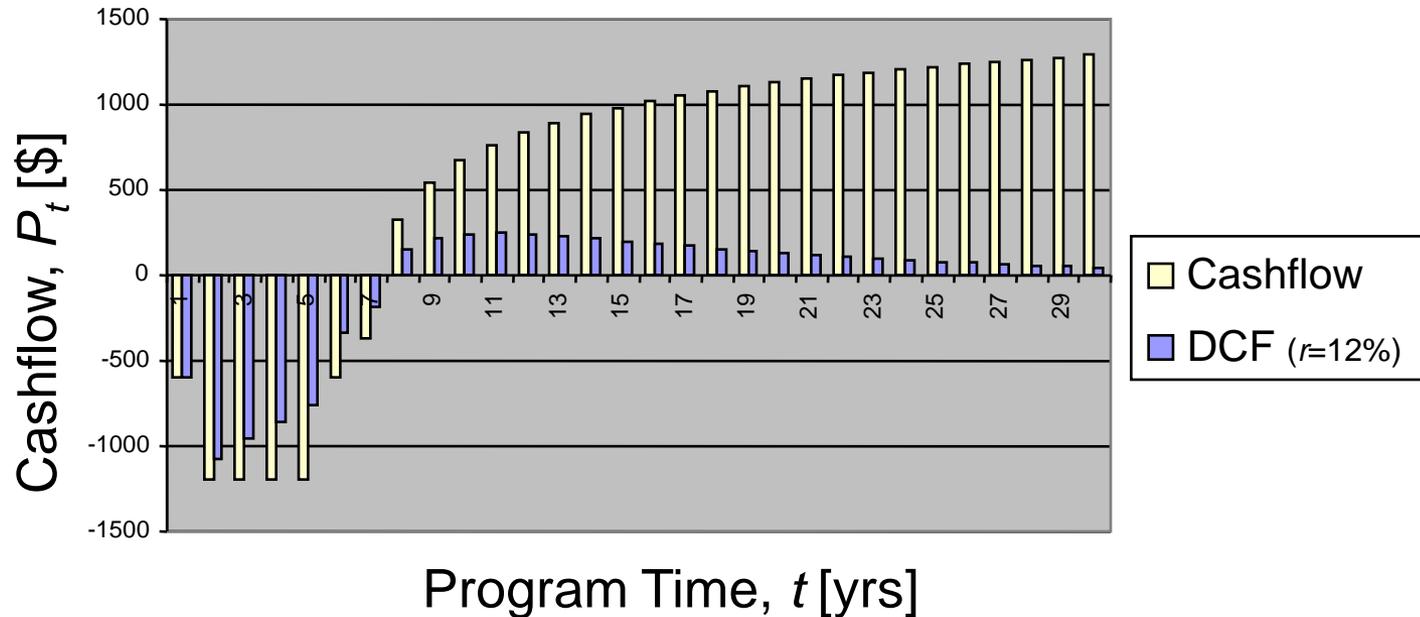
Period	Discount Factor	Cash Flow	Present Value
0	1	-150,000	-150,000
1	0.935	-100,000	-93,500
2	0.873	+300,000	+261,000

Discount rate = 7%

NPV = \$18,400

- DCF analysis assumes a fixed schedule of cash flows
- What about uncertainty?
- Common approach: use a risk-adjusted discount rate
- The discount rate is often used to reflect the risk associated with a project: the riskier the project, use a higher discount rate
- Typical discount rates for commercial aircraft programs: 12-20%
- Issues with this approach?

$$NPV = \sum_{t=0}^T \frac{C_t}{(1+r)^t}$$



- How long it takes before entire initial investment is recovered through revenue
- Insensitive to time value of money, i.e. no discounting
- Gives equal weight to cash flows before cut-off date & no weight to cash flows after cut-off date
- Cannot distinguish between projects with different NPV
- Difficult to decide on appropriate cut-off date

- Payback criterion modified to account for the time value of money
 - Cash flows before cut-off date are discounted
- Overcomes objection that equal weight is given to all flows before cut-off date
- Cash flows after cut-off date still not given any weight

- Investment criterion is “rate of return must be greater than the opportunity cost of capital”
- Internal rate of return is equal to the discount rate for which the NPV is equal to zero

$$NPV = C_0 + \frac{C_1}{1 + IRR} + \frac{C_2}{(1 + IRR)^2} + \dots + \frac{C_T}{(1 + IRR)^T} = 0$$

- IRR solution is not unique
 - Multiple rates of return for same project
- IRR doesn't always correlate with NPV
 - NPV does not always decrease as discount rate increases

- Return of an action divided by the cost of that action

$$ROI = \frac{\text{revenue} - \text{cost}}{\text{cost}}$$

- Need to decide whether to use actual or discounted cashflows

- NPV analysis with different future scenarios
- Weighted by probability of event occurring

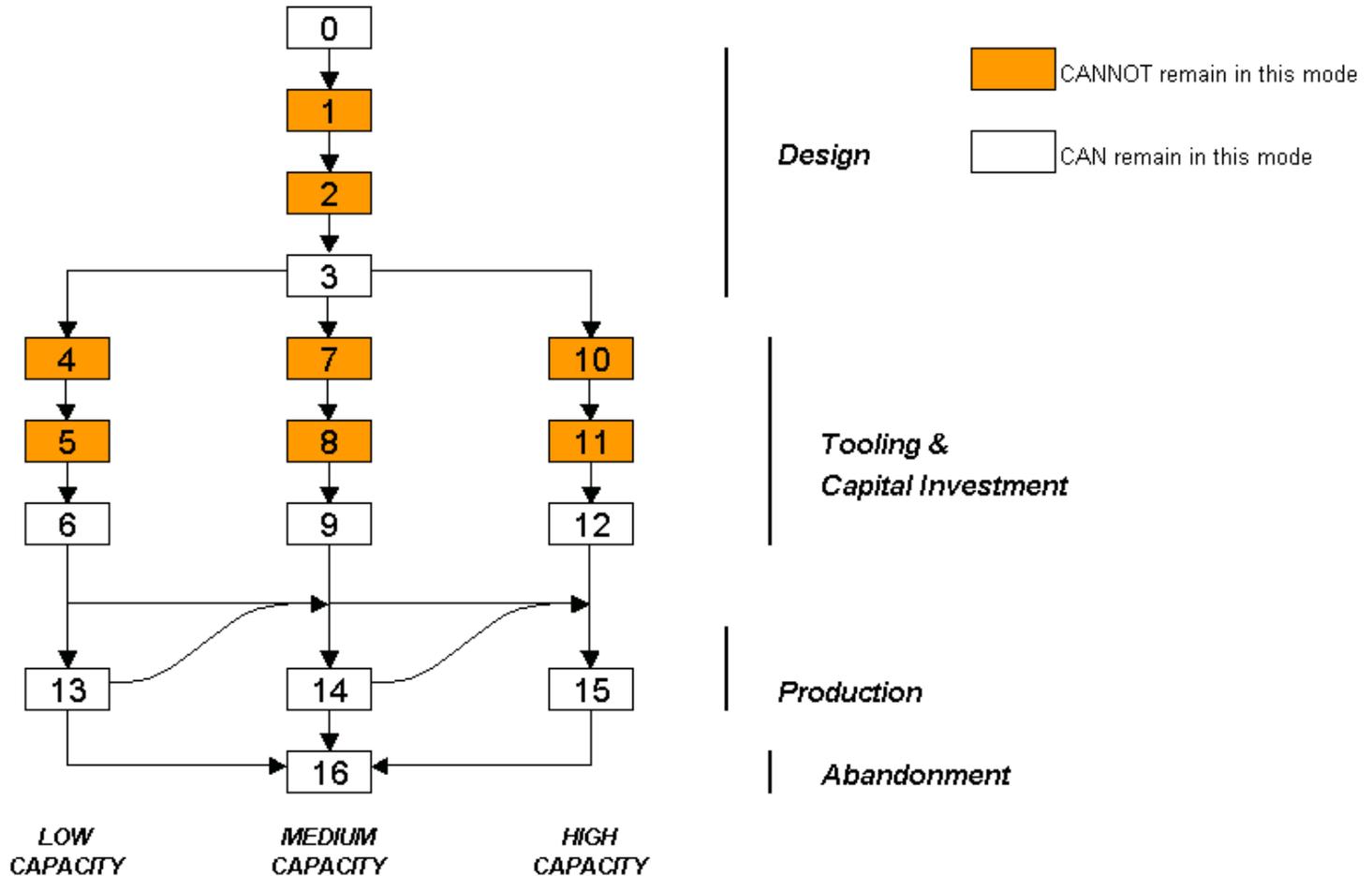
- In reality:
 - Cashflows are uncertain
 - Ability to make decisions as future unfolds
- View an aircraft program as a series of investment decisions
- Spending money on development today gives the option to build and sell aircraft at a later date
- Better valuation metric: **expected NPV** from dynamic programming algorithm (Markish, 2002)

- The firm:
 - Portfolio of designs
 - Sequential development phases
 - Decision making
- The market:
 - Sale price is steady
 - Quantity demanded is unpredictable
 - Units built = units demanded
- Problem objective:
 - Which aircraft to design?
 - Which aircraft to produce?
 - When?

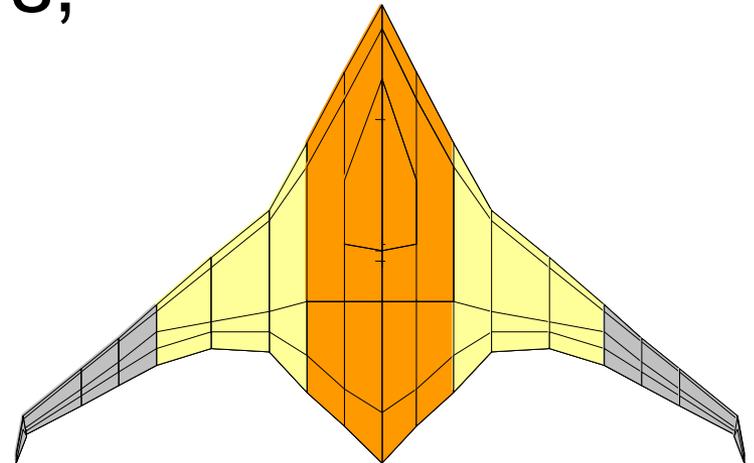
1. State variables s_t
2. Control variables u_t
3. Randomness
4. Profit function
5. Dynamics

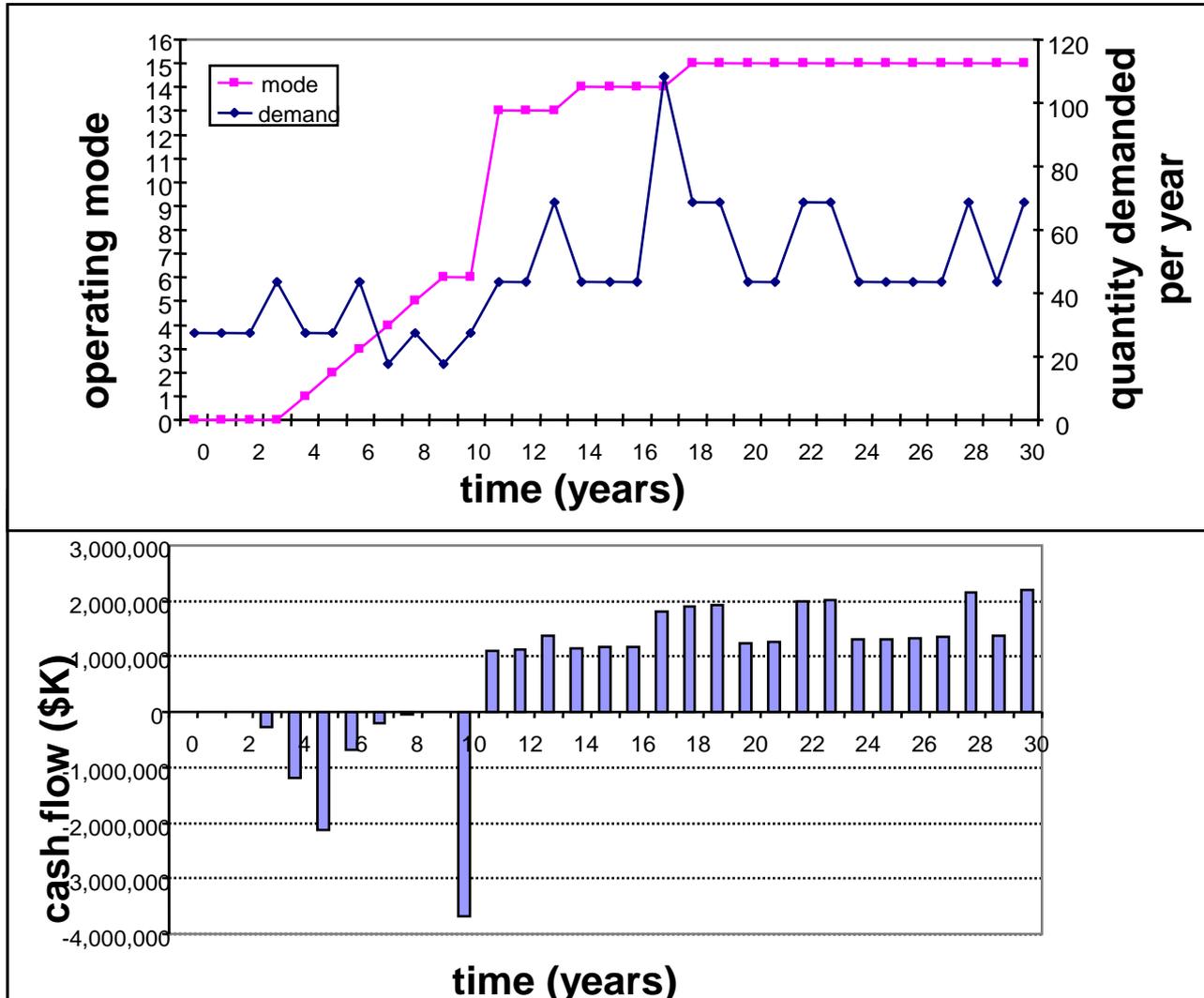
- Solution:
$$F_t(s_t) = \max_{u_t} \left\{ \pi_t(s_t, u_t) + \frac{1}{1+r} E_t[F_{t+1}(s_{t+1})] \right\}$$
- Solve recursively

How to model decision making?

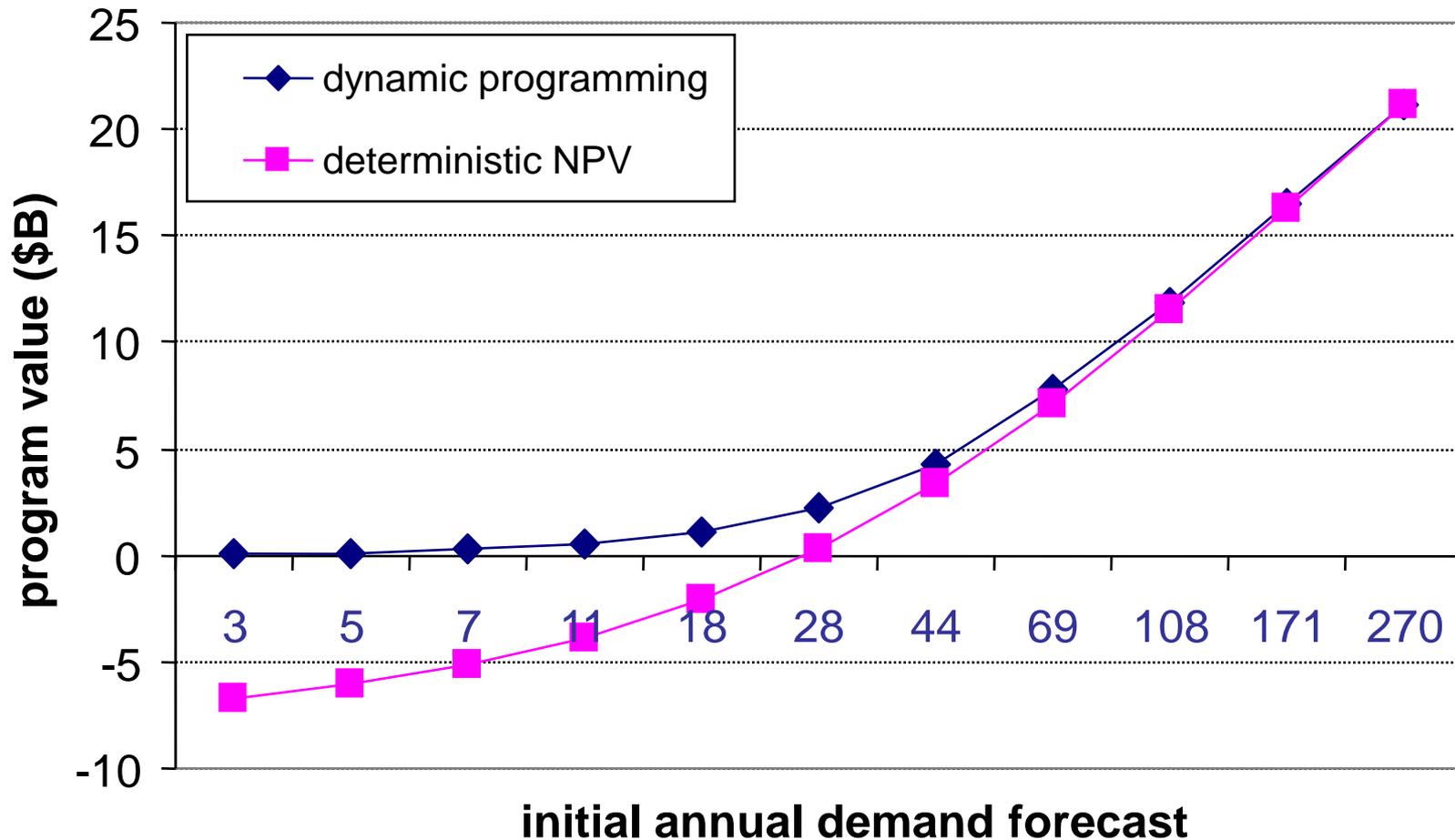


- Blended-Wing-Body (BWB):
 - Proposed new jet transport concept
- 250-seat, long range
- Part of a larger family sharing common centerbody bays, wings, ...



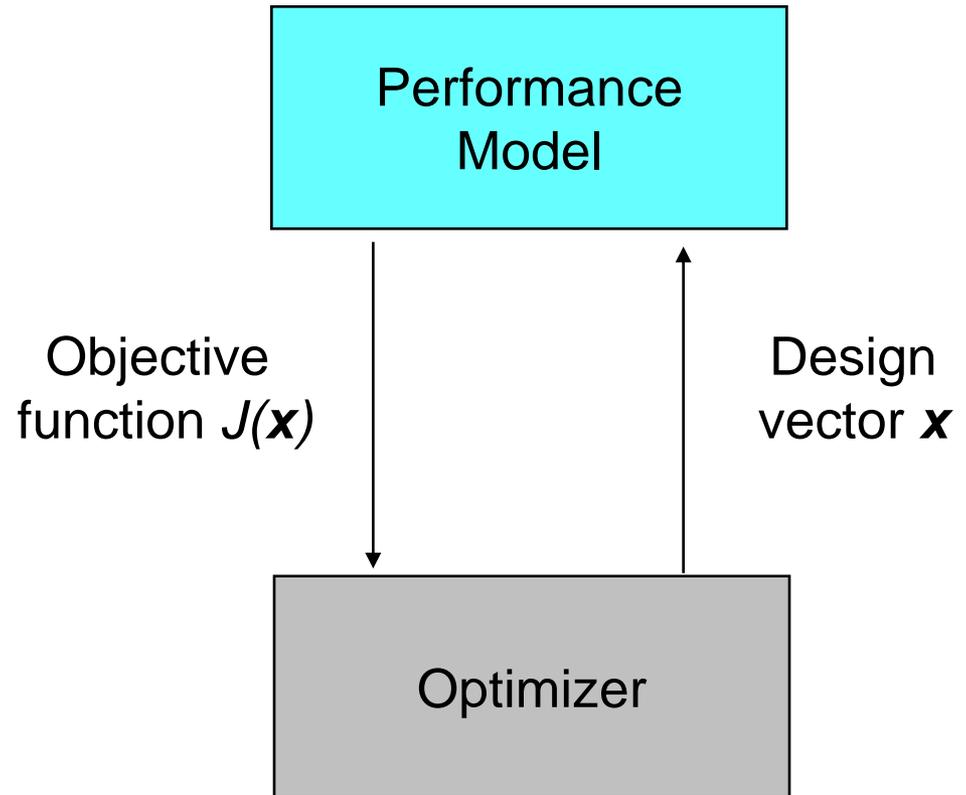


BWB Example: Importance of Flexibility

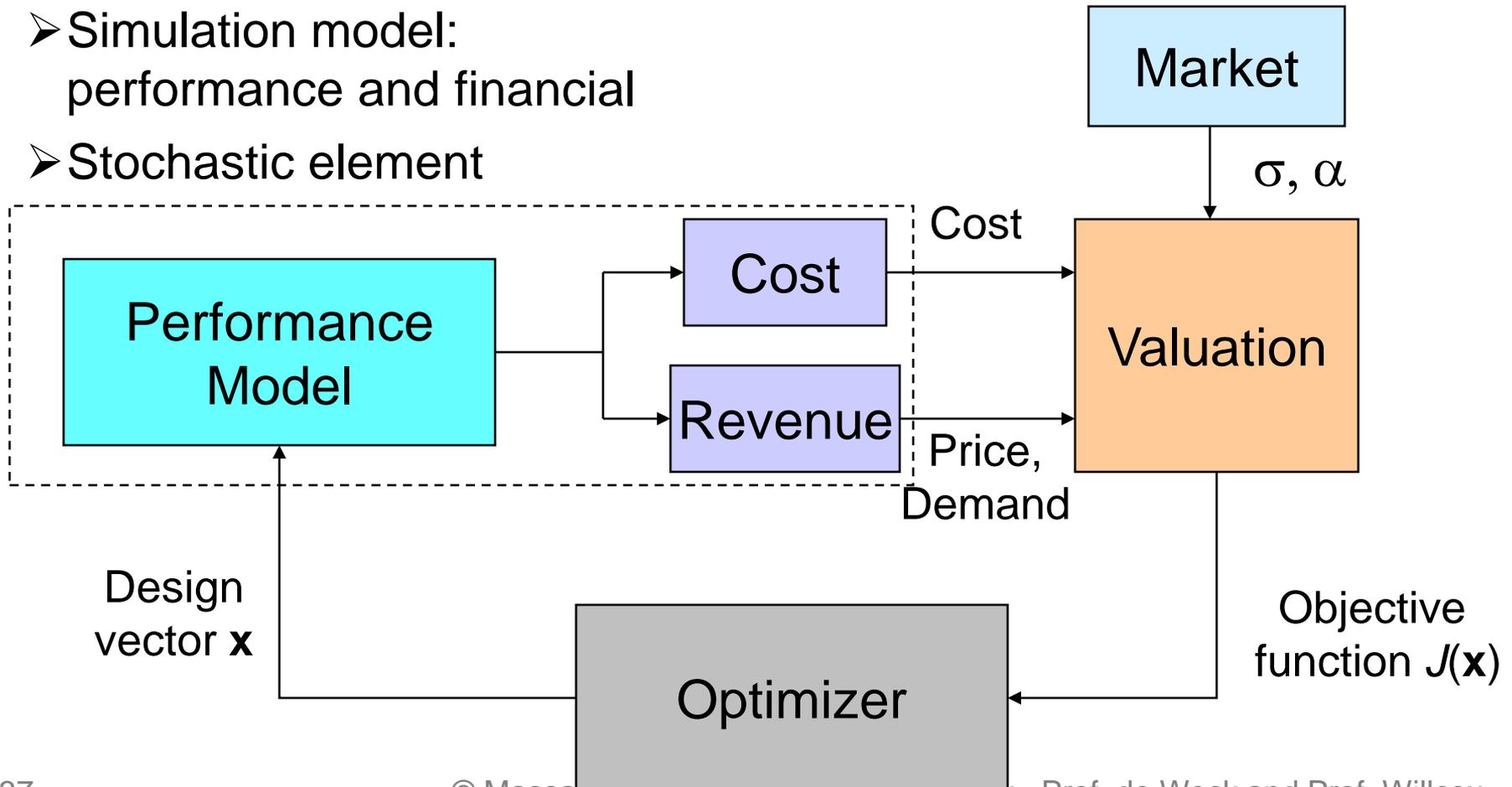


At baseline of 28 aircraft, DP value is \$2.26B versus deterministic value of \$325M

- Objective function: usually minimum weight
- Design vector: attributes of design, e.g. planform geometry
- Performance model: contains several engineering disciplines

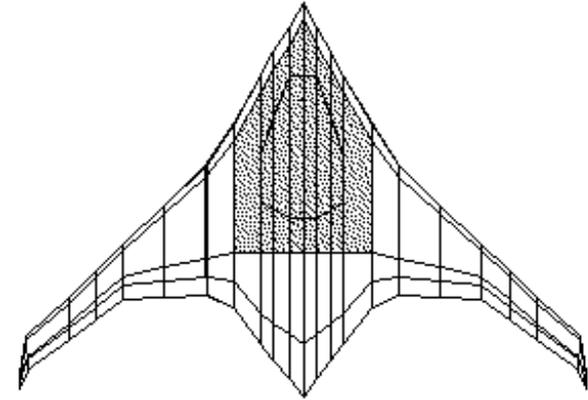


- Objective function: value metric, e.g. NPV
- Simulation model: performance and financial
- Stochastic element

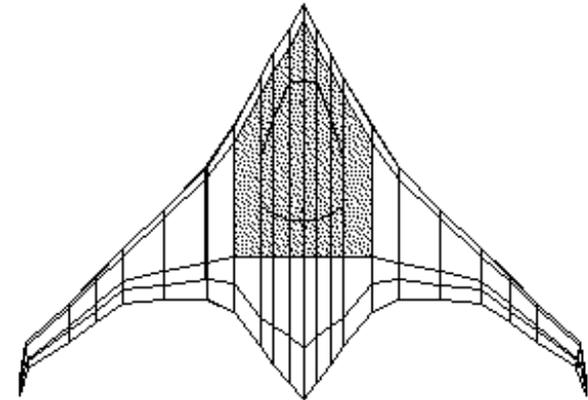


- Boeing BWB case study
 - 475 passengers, 7800 nmi range
 - Baseline: optimized for minimum GTOW
- Outcomes
 - Comparison of min GTOW and max $E[\text{NPV}]$ designs
 - Traditional NPV vs. stochastic $E[\text{NPV}]$
 - Effect of range requirement on program value
 - Effect of speed requirement on program value

- New objective results in tradeoff:
 - Lower structural weight, lower cost
 - Higher fuel burn, lower price
- Net result
 - 2.3% improvement in value
- Overall design very similar
 - Constrained to satisfy design requirements
 - Unable to move dramatically in design space

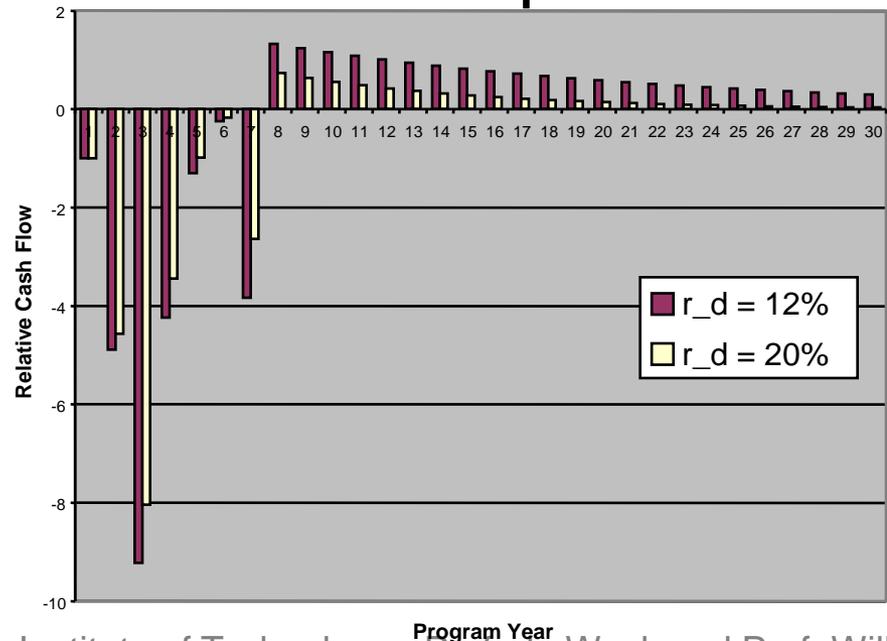


Minimum-GTOW planform

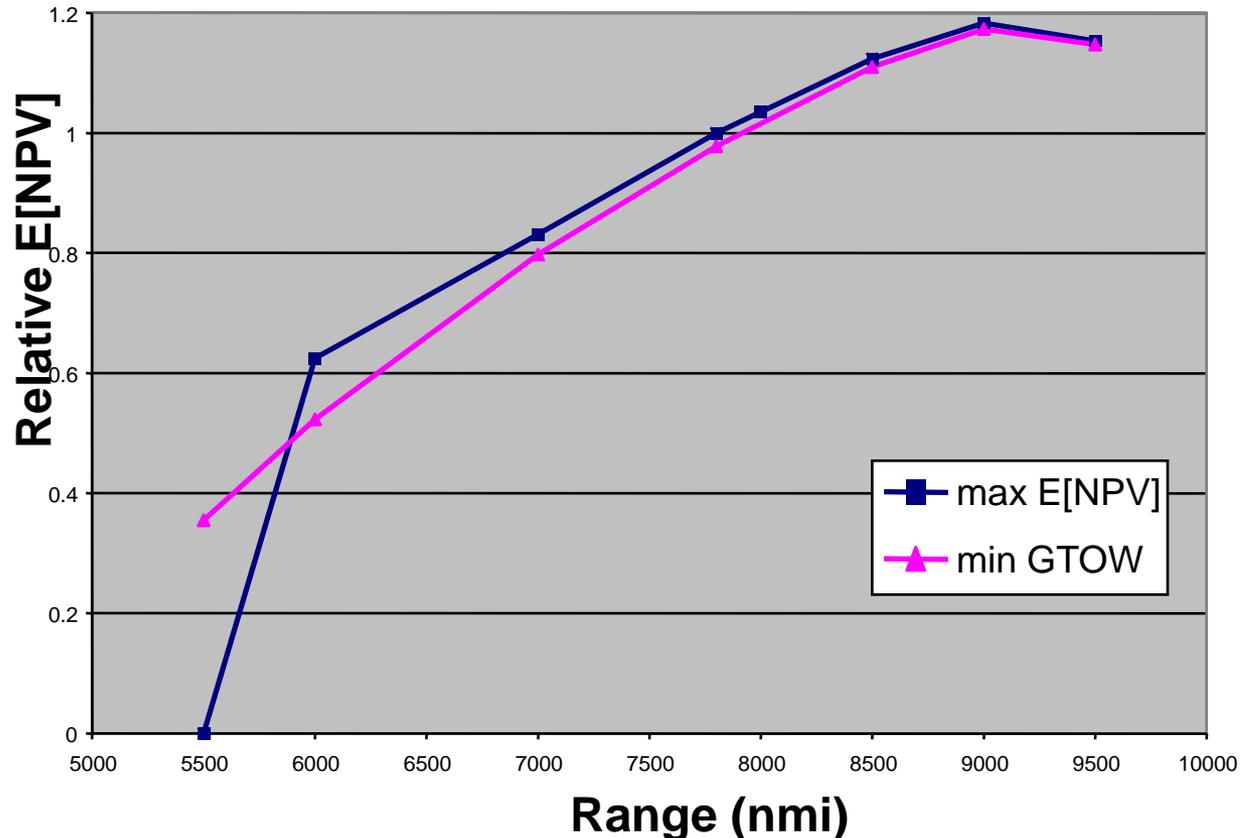


Maximum-value planform

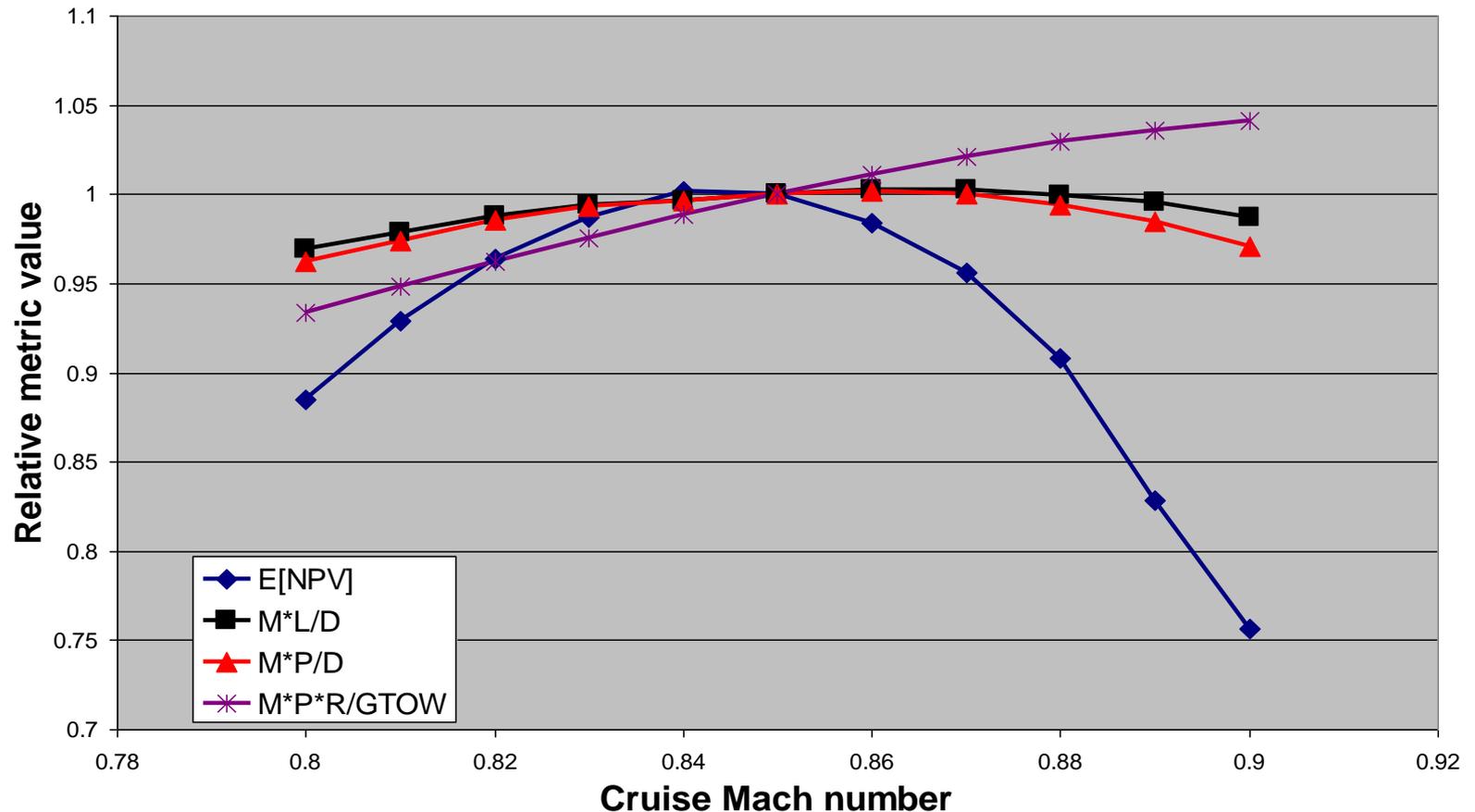
- Discount rates: 12% and 20%
 - Computational expense reduced, but NPV results are negative
 - E[NPV] for 12% design = 0.58% decrease
 - E[NPV] for 20% design = 3.7% decrease } relative to max-E[NPV] design
- High- r_d drives design to reduce development costs
- *Traditional NPV not appropriate*
 - As valuation metric
 - As optimization objective



- Comparison of min-GTOW and max-E[NPV] design solutions
- E[NPV] for varying ranges relative to max-E[NPV] design



- Comparison of E[NPV] and other metrics for varying speeds



- Designing for value is crucial: for a program to be successful, we cannot focus exclusively on performance
- The definition of value is flexible, and will vary depending on the application and on your interest as a stakeholder
- Financial metrics can be used to quantifying value, but use caution in your choice of value objective function
- Cost and revenue are difficult to model – often use empirical data
- It is important that uncertainty and risk are handled appropriately
- There is much work to be done on this issue

Brealey, R. & Myers, S., *Principles of Corporate Finance*, 7th Edition, McGraw-Hill, NY 2003

Markish, J., “Valuation Techniques for Commercial Aircraft Program Design”, Masters Thesis, MIT Dept. of Aeronautics & Astronautics, June 2002.

Markish, J. and Willcox, K., “Value-Based Multidisciplinary Techniques for Commercial Aircraft System Design”, *AIAA Journal*, Vol. 41, No. 10, October 2003, pp. 2004-12.

Peoples, R. and Willcox, K., “Value-Based Multidisciplinary Optimization for Commercial Aircraft Design and Business Risk Assessment,” *Journal of Aircraft*, Vol. 43, No. 4, July-August, 2006, pp. 913-921.

Roskam, J., *Airplane Design Part VIII*, 1990.

Raymer, D., *Aircraft Design: A Conceptual Approach*, 3rd edition, 1999.

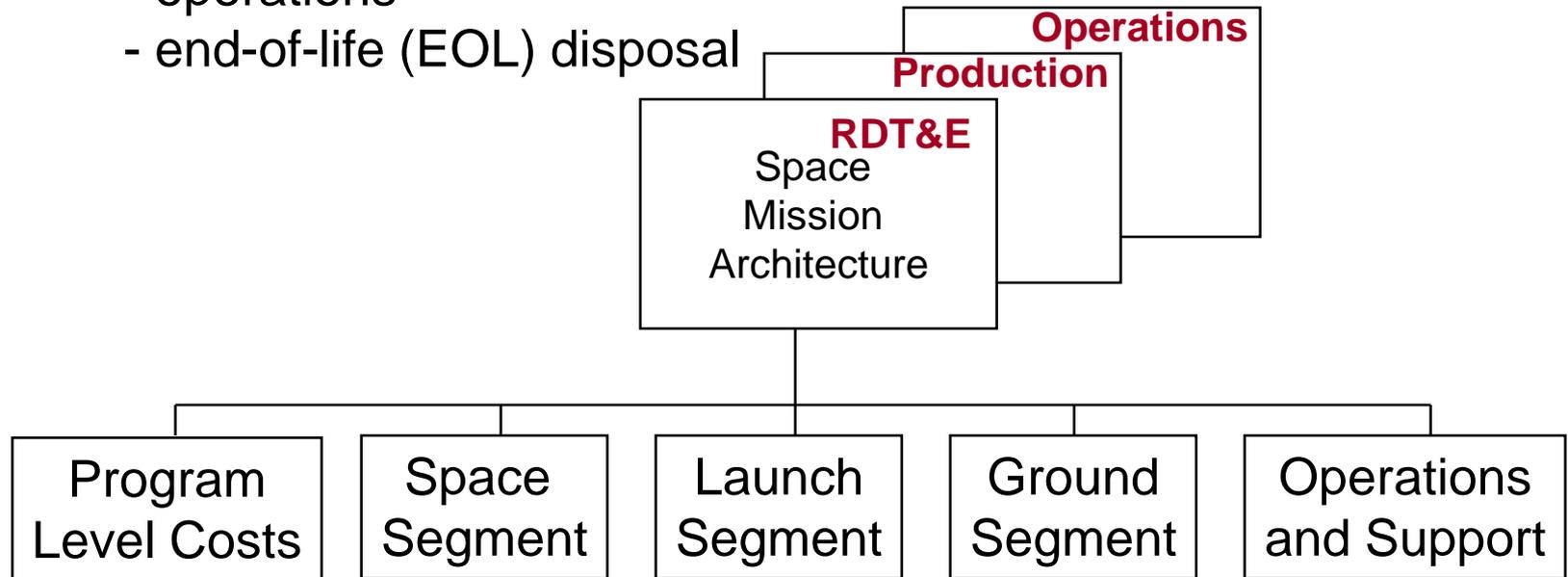
Schaufele, R., *The Elements of Aircraft Preliminary Design*, 2000.

This section will:

- present a process for obtaining system cost estimates, particularly for space systems
- provide cost-estimating relationships
- describe how to assess uncertainty in cost estimates
- provide a specific example: optical systems cost

Organizational Table that collects costs, covers:

- research, development, test and evaluation (RDT&E)
- production, including learning curve effects
- launch and deployment
- operations
- end-of-life (EOL) disposal



- Management
- Systems Eng
- Integration

- Payload
- Spacecraft
- Software
- “Systems”

- Launch Vhc
- Launch Ops
- S/C-L/V
- integration

- Facilities
- Equipment
- Software
- etc

- Personnel
- Training
- Maintenance
- Spares

Basic techniques to develop Cost Models:

(1) Detailed bottom-up estimating

- identify and specify lower level elements
- estimated cost of system is Σ of these
- time consuming, not appropriate early, accurate

(2) Analogous Estimating

- look at similar item/system as a baseline
- adjust to account for different size and complexity
- can be applied at different levels

(3) Parametric Estimating

- uses Cost Estimation Relationships (CER's)
- needed to find theoretical first unit (TFU) cost

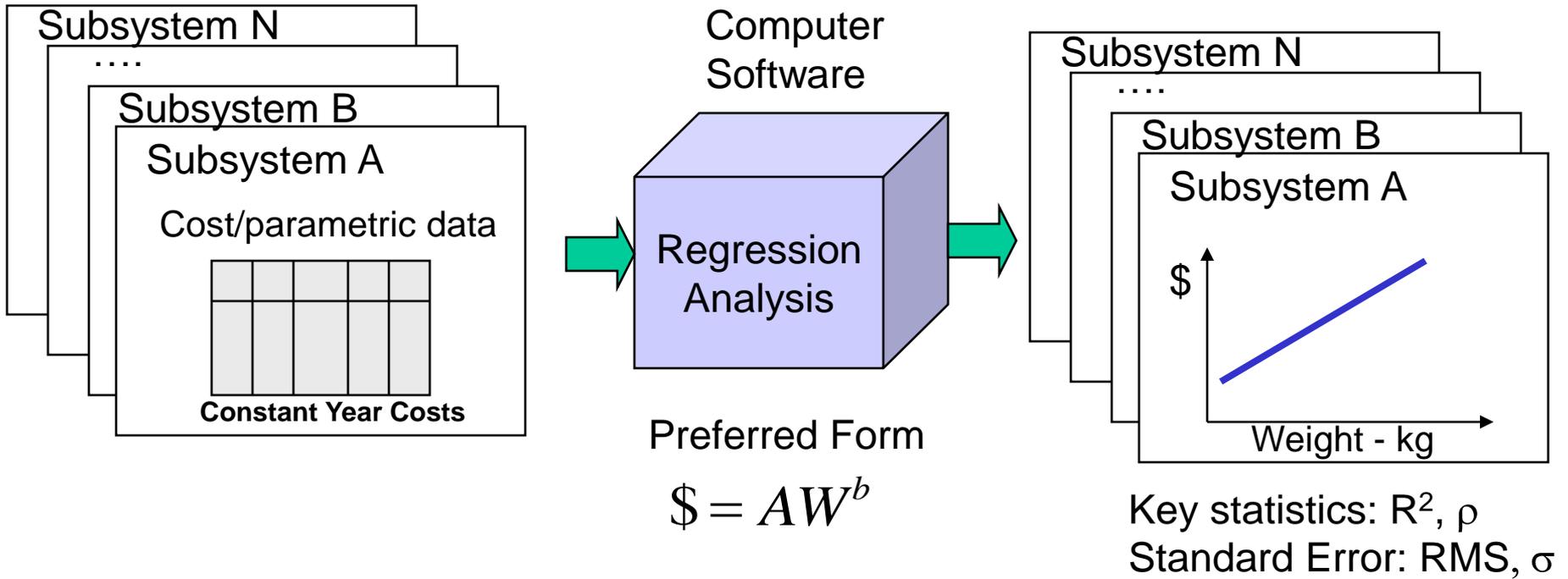
Are most appropriate for trade studies:

Advantages:

- less time consuming than traditional bottom-up estimates
- more effective in performing cost trades
- more consistent estimates
- traceable to specific class of (space) systems

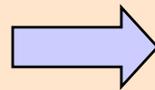
Major Limitations:

- applicable only to parametric range of historical data
- lacking new technology factors, adjust CER to account for new technology
- composed of different mix of “things” in element to be costed
- usually not accurate enough for a proposal bid



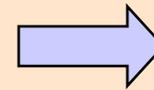
Step 1

Develop Database File



Step 2

Apply Regression Analysis



Step 3

Obtain CER's and Error Statistics

It is critical that cost estimated be based on a constant-year dollar bases. Reason: INFLATION

E.g. All costs are adjusted to FY92 (“Fiscal Year 1992”)

$$C_Y = R \cdot C_{Y-N}$$

Past Years

Use actual inflation numbers

$$R = \underbrace{1.040}_{FY92} \underbrace{1.037}_{FY93} \underbrace{1.034}_{FY94} = 1.115$$

Convert Oct-1991 cost to Oct-1994 costs

Future Years

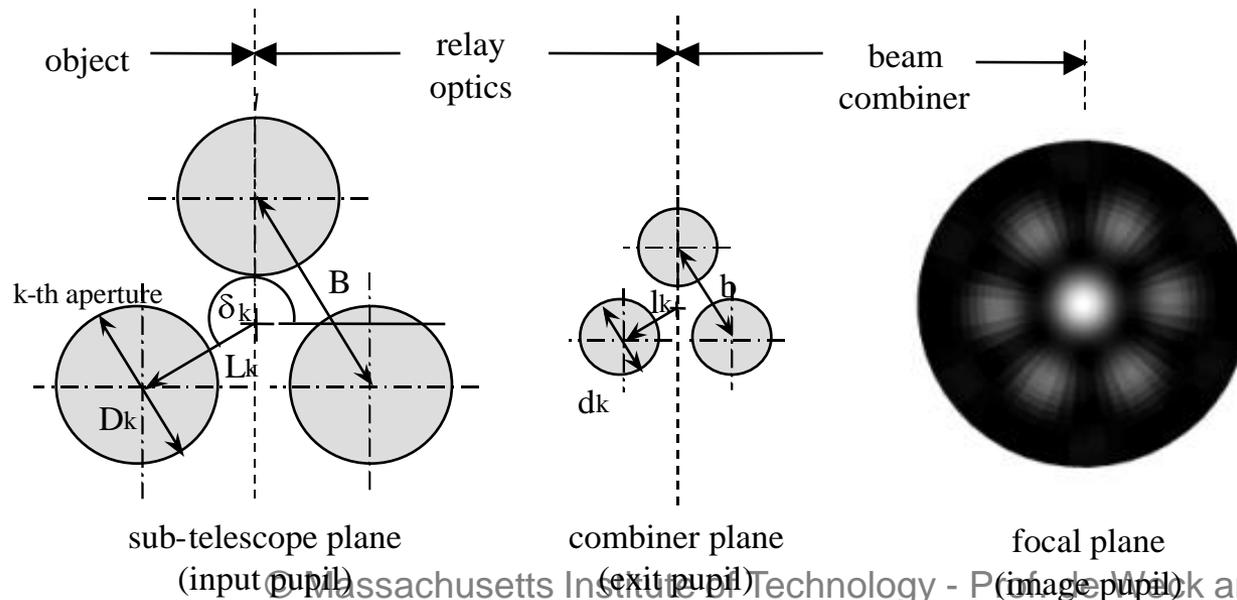
Use forecasted inflation numbers
e.g. 3.1% yearly inflation in U.S.

$$R = 1 + i_{RATE}^N$$

\$ 1M in FY 1980 corresponds to \$ 2.948M in FY 2005 (projected)

See Table 20-1 on handout

- Investigate economical viability of modular optics given performance constraints
- Focus on monolithic Cassegrain telescopes versus Goly-3 design
- Use real data and experience from ARGOS



Kahan, Targrove, “Cost modeling of large spaceborne optical systems”, SPIE, Kona, 1998

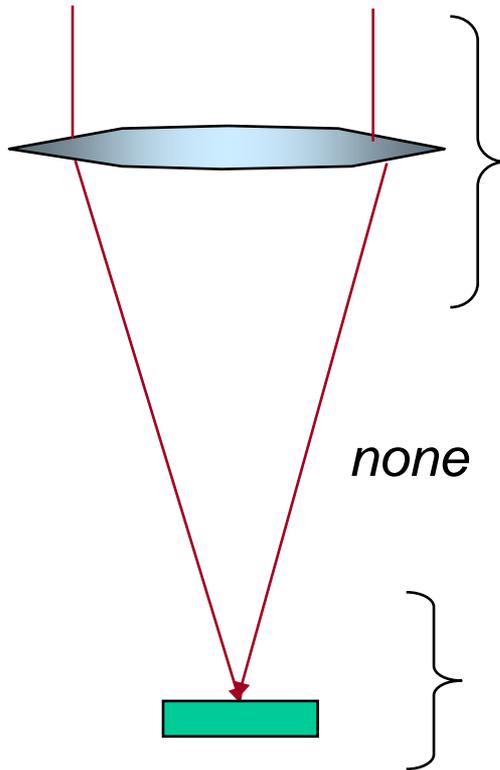
Humphries, Reddish, Walshaw, “Cost scaling laws and their origin: design strategy for an optical array telescope”, IAU, 1984

Meinel, “Cost-scaling laws applicable to very large optical telescopes”, SPIE, 1979

Meinel’s law: $S = 0.37 \cdot D^{2.58}$ [M\$] (1980)

Cost Modeling Approach

Monolithic System

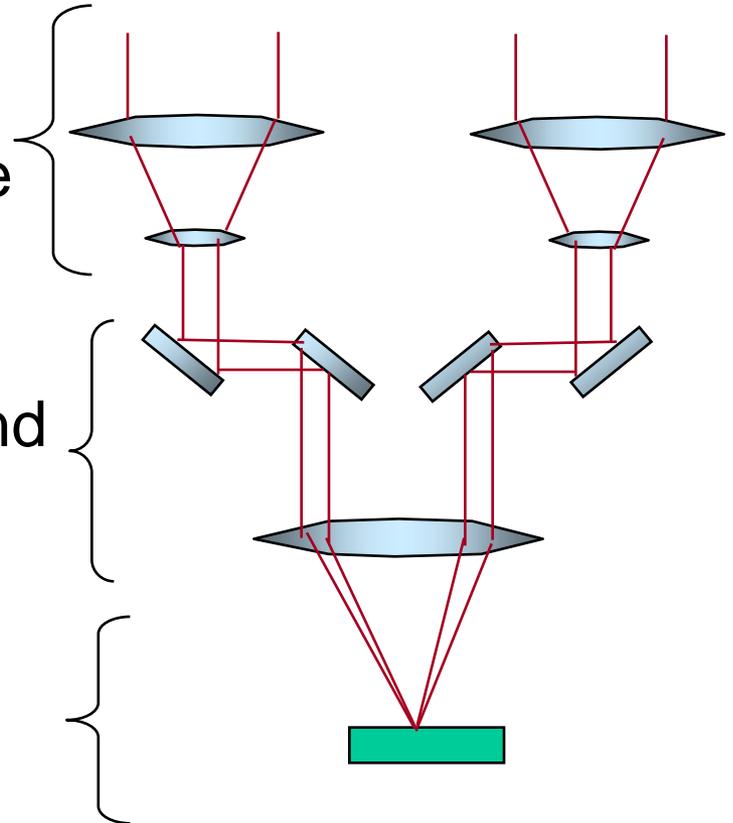


(A)
Sub-telescope

(B)
Relay optics and
combiner

(C)
Detector

Modular Golay System



$$\text{Total Cost: } \Sigma_{\text{total}} = C_A + C_B + C_C$$

Does not include
development cost

Cost Estimation Relationships

(A) Optical Telescope Assy: $C_A = L_A \cdot D^\alpha$

Modular array telescope:
(includes learning curve)

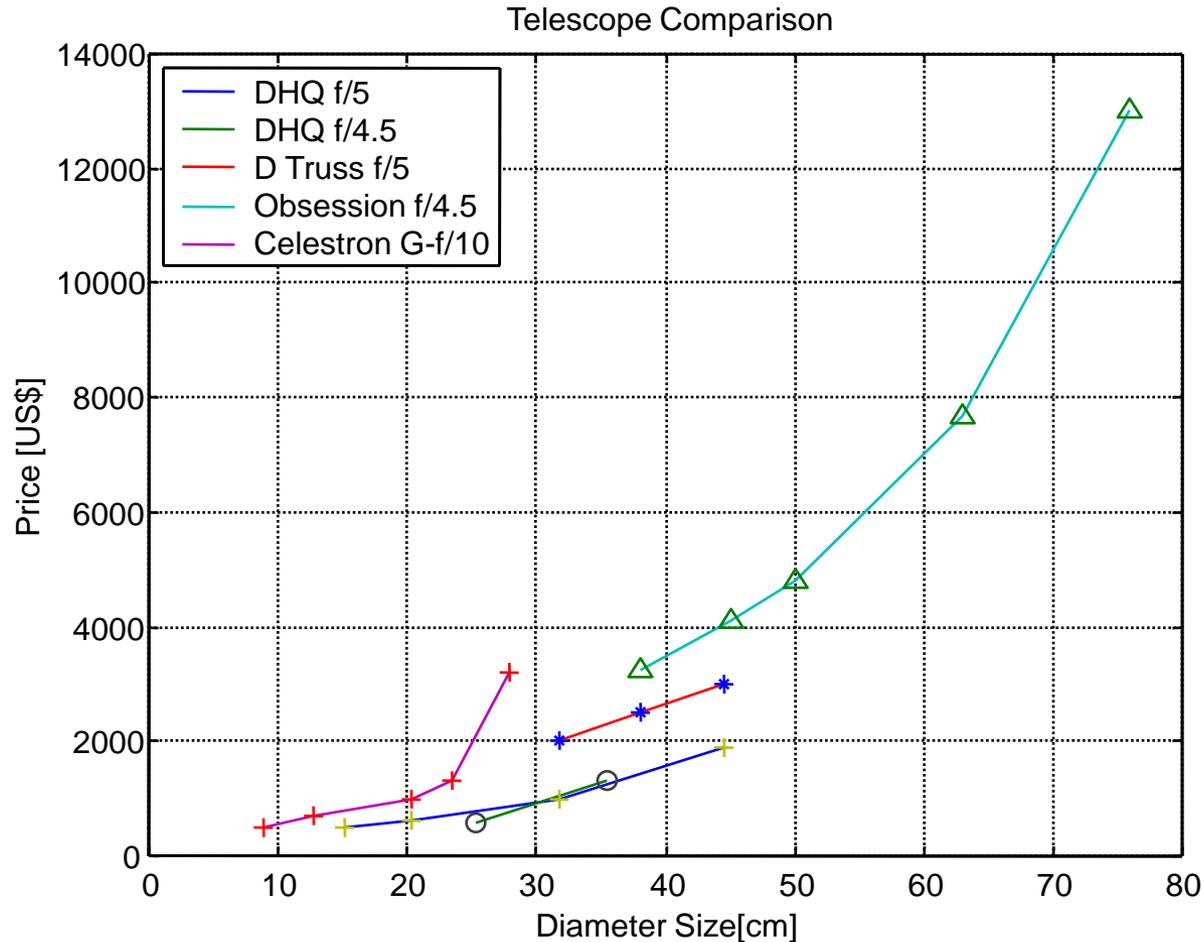
$$C_A = \underbrace{\left[L_A \cdot D^\alpha \right]}_{\text{TFU}} \cdot N^B$$

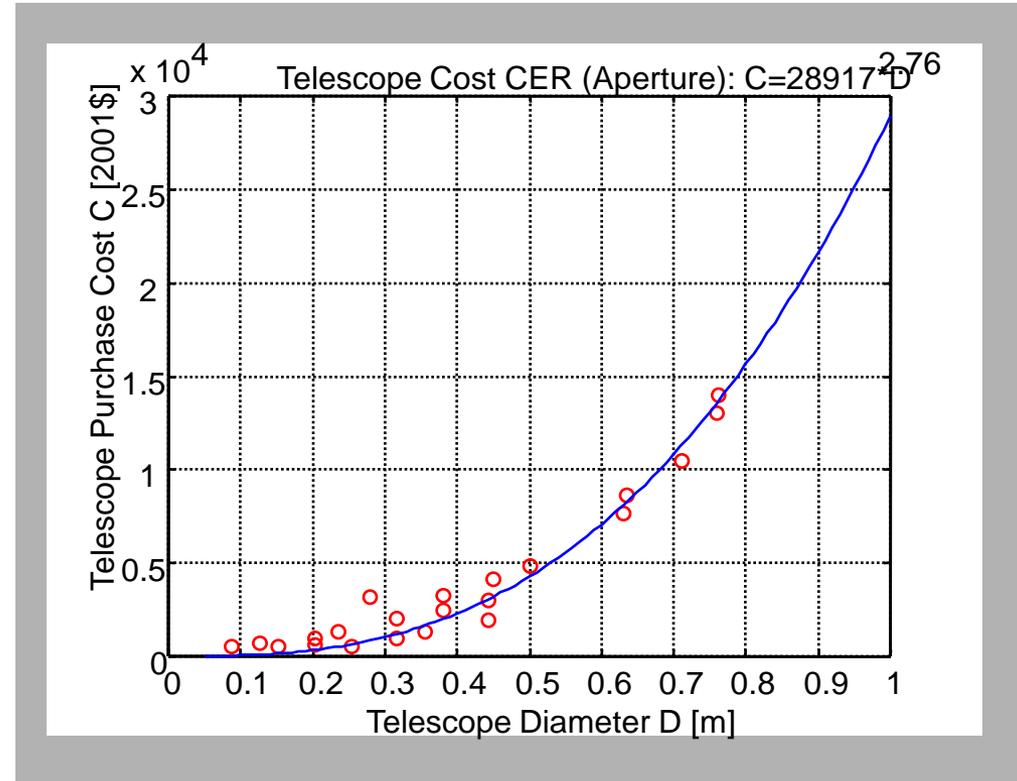
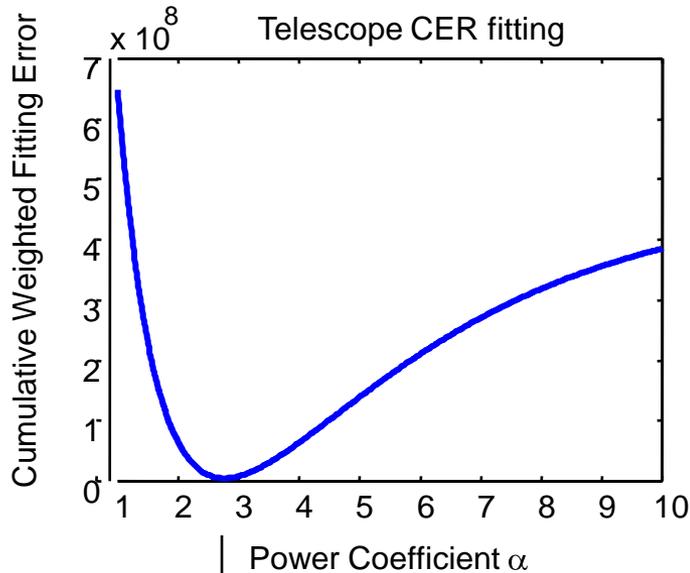
$$B = 1 - \left[\log 100\% / S / \log 2 \right]$$

(B) Relay & Combining Optics: Use ARGOS cost database

(C) Detector: CCD: $C_c = L_c \cdot \sqrt{n_{pix}}^\alpha$

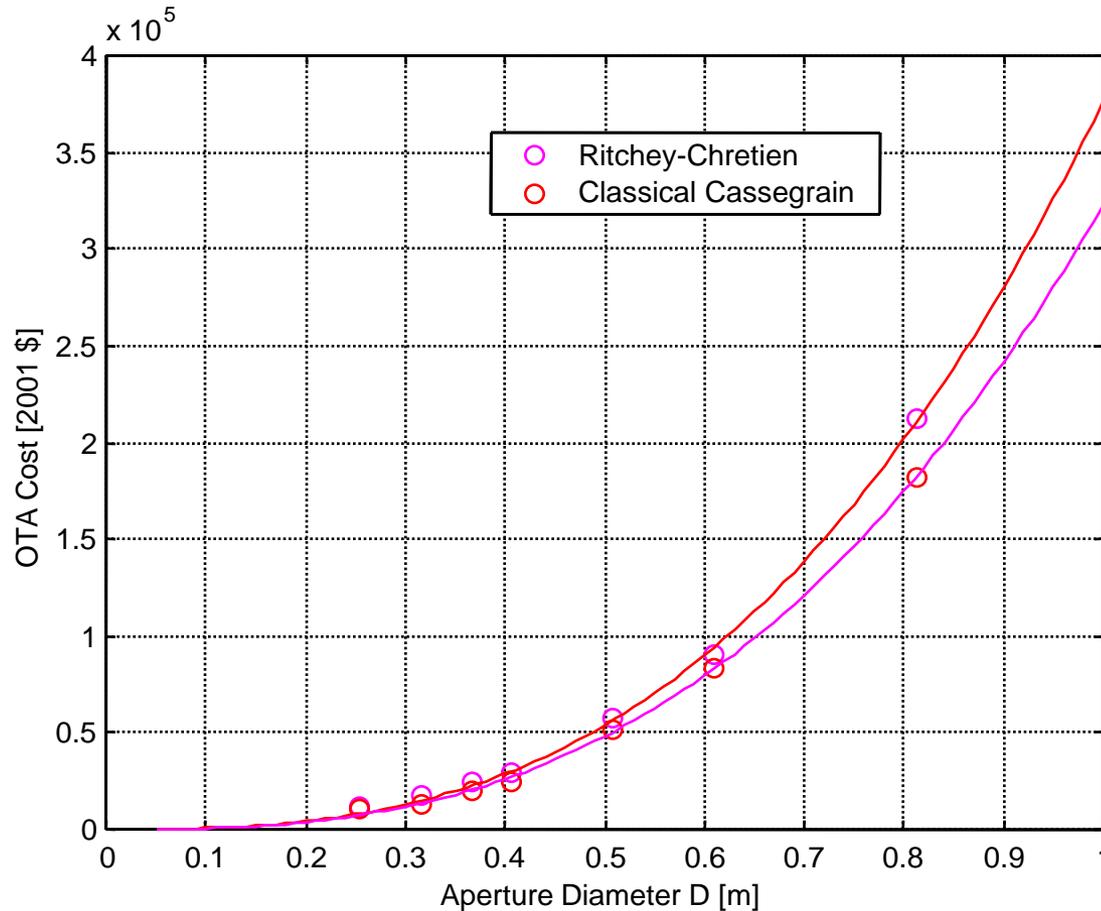
Next Step: Need to determine coefficients based on available commercial pricing data





Minimum yields best CER fit
exponent $\alpha=2.76$

Professional Telescope OTA cost



CERs for
Ritchey-Chretien

$$C_{RC} = 376000 \cdot D^{2.80}$$

Classical Cassegrain

$$C_{CC} = 322840 \cdot D^{2.75}$$

Remarkable Result:
virtually identical
power law across
completely different
product lines.

Company: Optical Guidance Systems

(<http://www.opticalguidancesystems.com>)

Cost of relay optics depends on:

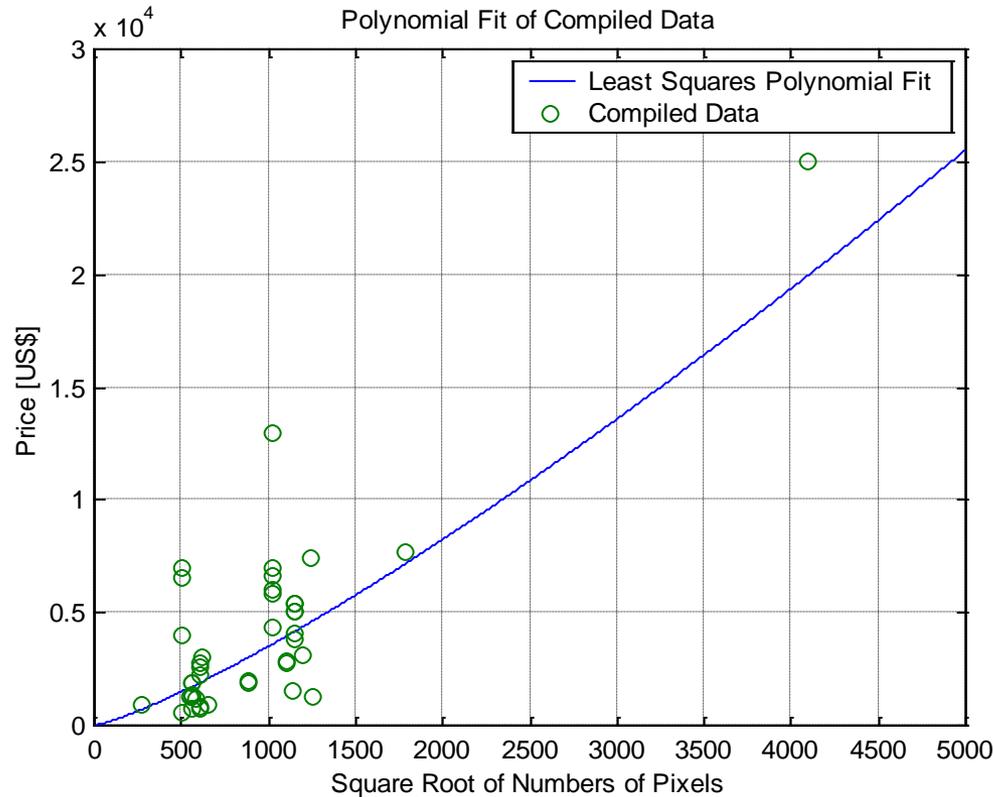
- number and type of actuators: ODL, FSM, active translation stages, active pyramid mirrors
- sensing elements (CCD, quad cells etc)
- aperture magnification m_a
- Quality requirements (RMS WFE)

As interim solution use ARGOS cost database:

Passive Optics: Collimators, Combiner, etc	\$ 32,874.-
Active Optics: FSM, Fold Mirrors etc...	\$ 18,859.-

$$m_a = 10$$

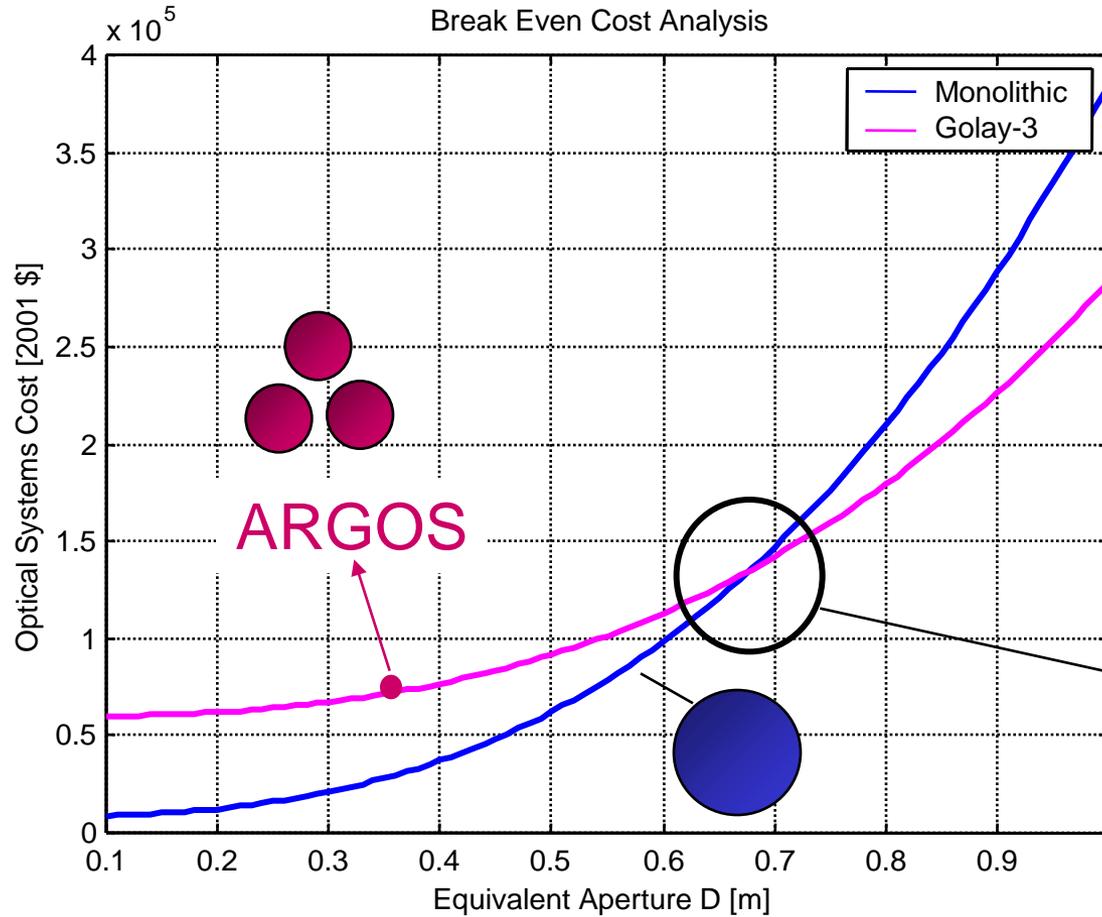
About: **\$ 50,000.-**



AP-9 (KAF-6303E)

$$C_{CCD} = 0.68 \cdot \sqrt{n_{pix}}^{1.23}$$

It appears that CCD cost also depends on factors other than raw pixel count. **Need to investigate.**



Assumptions:

$N=3$

$n_{\text{pix}} = 2048$

Relay optics
and combiner
costs fixed

Preliminary
Analysis
suggest
crossover
around 0.7 m

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ESD.77 / 16.888 Multidisciplinary System Design Optimization
Spring 2010

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