

# 1.050 Engineering Mechanics I

## Review session

1

## 1.050 – Content overview

### I. Dimensional analysis

1. On monsters, mice and mushrooms
2. Similarity relations: Important engineering tools

Lectures 1-3  
Sept.

### II. Stresses and strength

3. Stresses and equilibrium
4. Strength models (how to design structures, foundations.. against mechanical failure)

Lectures 4-15  
Sept./Oct.

### III. Deformation and strain

5. How strain gages work?
6. How to measure deformation in a 3D structure/material?

Lectures 16-19  
Oct.

### IV. Elasticity

7. Elasticity model – link stresses and deformation
8. Variational methods in elasticity

Lectures 20-32  
Oct./Nov.

### V. How things fail – and how to avoid it

9. Elastic instabilities
10. Fracture mechanics
11. Plasticity (permanent deformation)

Lectures 33-37  
Dec.

2

## Notes regarding final exam

- Please contact me or stop by at any time for any questions
- The final will be comprehensive and cover **all material discussed in 1.050**. Note that the last two p-sets will be important for the final.
  - To get an idea about the style of the final, work out old finals and the practice final
  - There will be 2-3 problems with several questions each (e.g. beam problem/ truss problem, continuum problem)
  - We will post old final exams from 2005 and 2006 today
  - We will post an additional, new practice final exam on or around Wednesday next week
  - Another list of variables and concepts will be posted next week
  - **Stay calm, read carefully, and practice time management**

3

## Stress, strain and elasticity - concepts

4

## Overview: 3D linear elasticity

**Stress tensor**  $\underline{\underline{\sigma}}(\vec{x})$

**Basis: Physical laws**  
(Newton's laws)

$$\left. \begin{array}{l} \text{Statically admissible (S.A.)} \\ \left\{ \begin{array}{l} \text{BCs on boundary of domain } \Omega \\ \partial\Omega_{\vec{T}^d} : \vec{T}^d(\vec{n}) = \underline{\underline{\sigma}} \cdot \vec{n} \\ \Omega : \left\{ \begin{array}{l} \vec{T}(\vec{n}) = \underline{\underline{\sigma}} \cdot \vec{n} \\ \text{div } \underline{\underline{\sigma}} + \rho\vec{g} = 0 \\ \sigma_{ij} = \sigma_{ji} \end{array} \right. \end{array} \right. \end{array} \right\}$$

**Strain tensor**  $\underline{\underline{\varepsilon}}(\vec{x})$

**Basis: Geometry**

$$\left. \begin{array}{l} \text{Kinematically admissible (K.A.)} \\ \left\{ \begin{array}{l} \text{BCs on boundary of domain } \Omega \\ \partial\Omega_{\vec{\xi}^d} : \vec{\xi}^d = \vec{\xi} \\ \text{Linear deformation theory} \\ \|\text{Grad } \vec{\xi}\| \ll 1 \\ \underline{\underline{\varepsilon}} = \frac{1}{2} \left( \text{grad } \vec{\xi} + (\text{grad } \vec{\xi})^T \right) \\ \varepsilon_{ij} = \frac{1}{2} \left( \frac{\partial \xi_i}{\partial x_j} + \frac{\partial \xi_j}{\partial x_i} \right) \end{array} \right. \end{array} \right\}$$

**Elasticity** **Basis: Thermodynamics**

$$\underline{\underline{\sigma}} = \underline{\underline{c}} : \underline{\underline{\varepsilon}} \quad \sigma_{ij} = c_{ijkl} \varepsilon_{kl} \quad \underline{\underline{\sigma}} = \left( K - \frac{2}{3} G \right) \varepsilon_v \underline{\underline{1}} + 2G \underline{\underline{\varepsilon}}$$

Isotropic solid

5

## Isotropic elasticity

$$\underline{\underline{\sigma}} = \left( K - \frac{2}{3} G \right) \varepsilon_v \underline{\underline{1}} + 2G \underline{\underline{\varepsilon}} = \left( K - \frac{2}{3} G \right) (\varepsilon_{11} + \varepsilon_{22} + \varepsilon_{33}) \underline{\underline{1}} + 2G \underline{\underline{\varepsilon}}$$

$$\left\{ \begin{array}{l} \sigma_{11} = \left( K - \frac{2}{3} G \right) (\varepsilon_{11} + \varepsilon_{22} + \varepsilon_{33}) + 2G \varepsilon_{11} \\ \sigma_{22} = \left( K - \frac{2}{3} G \right) (\varepsilon_{11} + \varepsilon_{22} + \varepsilon_{33}) + 2G \varepsilon_{22} \\ \sigma_{33} = \left( K - \frac{2}{3} G \right) (\varepsilon_{11} + \varepsilon_{22} + \varepsilon_{33}) + 2G \varepsilon_{33} \\ \sigma_{12} = 2G \varepsilon_{12} \\ \sigma_{23} = 2G \varepsilon_{23} \\ \sigma_{13} = 2G \varepsilon_{13} \\ \sigma_{11} = \left( K + \frac{4}{3} G \right) \varepsilon_{11} + \left( K - \frac{2}{3} G \right) \varepsilon_{22} + \left( K - \frac{2}{3} G \right) \varepsilon_{33} \\ \sigma_{22} = \left( K - \frac{2}{3} G \right) \varepsilon_{11} + \left( K + \frac{4}{3} G \right) \varepsilon_{22} + \left( K - \frac{2}{3} G \right) \varepsilon_{33} \\ \sigma_{33} = \left( K - \frac{2}{3} G \right) \varepsilon_{11} + \left( K - \frac{2}{3} G \right) \varepsilon_{22} + \left( K + \frac{4}{3} G \right) \varepsilon_{33} \\ \sigma_{12} = 2G \varepsilon_{12} \\ \sigma_{23} = 2G \varepsilon_{23} \\ \sigma_{13} = 2G \varepsilon_{13} \end{array} \right.$$

Linear isotropic elasticity

Written out for individual stress tensor coefficients

Linear isotropic elasticity

Written out for individual stress tensor coefficients, collect terms that multiply strain tensor coefficients

$$\begin{aligned} c_{1111} = c_{2222} = c_{3333} &= K + \frac{4}{3} G \\ c_{1122} = c_{1133} = c_{2233} &= K - \frac{2}{3} G \\ c_{1212} = c_{2323} = c_{1313} &= 2G \end{aligned}$$

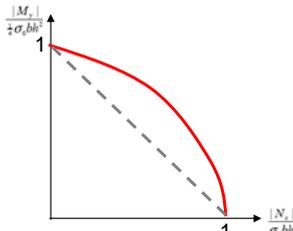
6

Variable	Definition	Notes & comments
$\nu$	$\varepsilon_{yy} = \varepsilon_{zz} = -\nu\varepsilon_{xx} \quad \nu = \frac{1}{2} \frac{3K - 2G}{3K + G}$	Poisson's ratio (lateral contraction under uniaxial tension)
$E$	$E = \frac{9KG}{3K + G}$ $\sigma_{xx} = E\varepsilon_{xx}$	Young's modulus (relates stresses and strains under uniaxial tension)
		Uniaxial beam deformation
		7

## Solving problems with strength approach

Use conditions for S.A. plus strength criterion (S.C.)

Variable	Definition	Notes & comments
Two pillars of stress-strength approach	At any point, $\underline{\sigma}$ must be: (1) Statically admissible (S.A.) <b>and</b> (2) Strength compatible (S.C.)	
	<ul style="list-style-type: none"> <li>Equilibrium conditions “only” specify statically admissible stress field, without worrying about if the stresses can actually be sustained by the material – <b>S.A.</b> <i>From EQ condition for a REV we can integrate up (upscale) to the structural scale</i> <b>Examples:</b> Many integrations in homework and in class; Hoover dam etc.</li> <li>Strength compatibility adds the condition that in addition to S.A., the stress field must be compatible with the strength capacity of the material – <b>S.C.</b> <i>In other words, at no point in the domain can the stress vector exceed the strength capacity of the material</i> <b>Examples:</b> Sand pile, foundation etc. – Mohr circle <sub>9</sub></li> </ul>	

Variable	Definition	Notes & comments
$D_S$	$\forall x; (\bar{F}_S, \bar{M}_S) \in D_S(x) \Leftrightarrow f(x, \bar{F}_S(x), \bar{M}_S(x)) \leq 0$	Strength domain for beams
$ M_y _{\text{lim}} = M_0$	$ M_y _{\text{lim}} = M_0 = \frac{1}{4} \sigma_0 b h^2$ For rectangular cross-section $b, h$	Moment capacity for beams
$ N_x _{\text{lim}} = N_0$	$ N_x _{\text{lim}} = N_0 = b h \sigma_0$	Strength capacity for beams
$f(M_y, N_x) \leq 0$	$f(M_y, N_x) = \frac{ M_y }{M_0} + \frac{ N_x }{N_0} - 1 \leq 0$ $f(M_y, N_x) = \frac{ M_y }{M_0} + \left( \frac{ N_x }{N_0} \right)^2 - 1 \leq 0$ 	M-N interaction (linear)  M-N interaction (actual); convexity

Variable	Definition	Notes & comments
Safe strength domain	$\forall i; 0 \leq Q^{(i)} \leq \alpha_i Q_{\text{lim}}^{(i)}$ $\sum_{i=1,n} \frac{Q^{(i)}}{Q_{\text{lim}}^{(i)}} \leq \sum_{i=1,n} \alpha_i = 1$ <p><math>Q_{\text{lim}}^{(i)}</math> : load bearing capacity of <math>i</math>-th load case</p>	Linear combination is safe (convexity)
		11

## Mohr circle

Display 3D strain tensor in 2D projection – enables us to ‘see’ largest shear stresses, largest normal stresses...

Thereby facilitates application of strength criterion

12

Variable	Definition	Notes & comments
$D_k$	$\forall \vec{x}; \sigma(\vec{x}) \in D_k \Leftrightarrow f(\vec{x}, \sigma(\vec{x})) \stackrel{\text{S.C.}}{\leq} 0$	Strength domain (general definition) Equivalent to condition for S.C.
$D_{k,\text{Tresca}}$	<p>Tresca: <math>\forall \vec{n}; f(\vec{T}) =  \tau  - c \leq 0</math></p> <p>cohesion, <math>2c = \sigma_0</math></p>	Tresca criterion
$D_{k,\text{Tension-cutoff}}$	<p><math>\forall \vec{n}; f(\vec{T}) = \sigma - c \leq 0</math></p> <p>Max. tensile stress</p>	Tension cutoff criterion

13

Variable	Definition	Notes & comments
$D_{k,\text{Mohr-Coulomb}}$	<p>Mohr-Coulomb: <math>\forall \vec{n}; f(\vec{T}) =  \tau  + \sigma \tan \varphi - c \leq 0</math></p>	Mohr-Coulomb
$\alpha_{\text{lim}}$	<p><math>\alpha_{\text{lim}} = \varphi</math></p>	Angle of repose

14

Variable	Definition	Notes & comments
$S$	$S = \int_S dS$	Cross-sectional area
$I$	$I = \int_S z^2 dS$	Second order area moment
$EI$	$M_y = -EI \frac{d^2 \xi_z^0}{dx^2} = EI \vartheta_y$	Beam bending stiffness (relates bending moment and curvature)
	$\frac{d^2 \xi_x^0}{dx^2} = -\frac{f_x}{ES}$	Governing differential equation, axial forces
	$\frac{d^4 \xi_z^0}{dx^4} = \frac{f_z}{EI}$	Governing differential equation, shear forces
	<ul style="list-style-type: none"> <li>• <b>Step 1:</b> Write down BCs (stress BCs and displacement BCs), analyze the problem to be solved (read carefully!)</li> <li>• <b>Step 2:</b> Write governing equations for <math>\xi_z, \xi_x, \dots</math></li> <li>• <b>Step 3:</b> Solve governing equations (e.g. by integration), results in expression with unknown integration constants</li> <li>• <b>Step 4:</b> Apply BCs (determine integration constants)</li> </ul>	Solution procedure to solve beam elasticity problems  15

## Energy approach

Approximate solution or find exact solution

16

# Upper/lower bounds

$$-\varepsilon_{\text{com}}(\underline{\underline{\sigma}}') \leq \left\{ \begin{array}{l} \max_{\underline{\underline{\sigma}}' \text{ S.A.}} (-\varepsilon_{\text{com}}(\underline{\underline{\sigma}}')) \\ \text{is equal to} \\ \min_{\vec{\xi}' \text{ K.A.}} \varepsilon_{\text{pot}}(\vec{\xi}') \end{array} \right\} \leq \varepsilon_{\text{pot}}(\vec{\xi}')$$

Lower bound

Solution

Upper bound

Complementary energy  
approach

Potential energy  
approach

$$\varepsilon_{\text{com}}(\underline{\underline{\sigma}}') = \psi^*(\underline{\underline{\sigma}}') - W^*(\vec{T}')$$

$$\varepsilon_{\text{pot}}(\vec{\xi}') = \psi(\underline{\underline{\varepsilon}}') - W(\vec{\xi}')$$

17

# Boundary conditions

Important concept in 1.050 and  
elsewhere

18

## Important BCs in beams/frames

Free end



$$\vec{F} = 0$$

$$\vec{M} = 0$$



$$\xi_z = 0$$

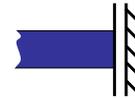
$$M_y = 0$$

Concentrated force



$$Q_z = -P$$

$P$



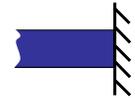
$$\xi_x = 0$$

$$\omega_y = 0$$

Hinge (bending)



$$M_y = 0$$



$$\xi_z = 0$$

$$\omega_y = 0$$

19

## Buckling of beams in compression

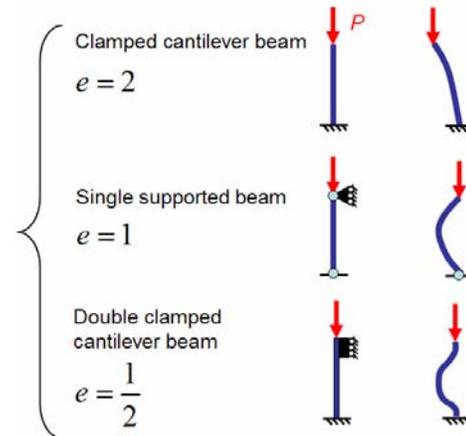
Elastic instability

20

# Buckling

$$P < P_{crit} = \frac{\pi^2 EI}{(el)^2}$$

$el$  "effective length"



Euler beam buckling  
Different boundary conditions

21

# Fracture mechanics

How to treat cracks in a continuum

22

# Example: 3D fracture model

Expressions for G can be found for a variety of geometries and structures:

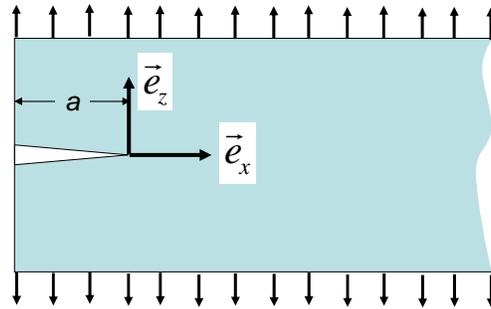
For this geometry:

$$G = 1.12^2 \frac{\pi a \sigma_0^2}{E} \stackrel{!}{=} 2\gamma_s$$

$$\sigma_0 = \sqrt{\frac{2\gamma_s E}{1.12^2 \pi a}}$$

$$\sigma_0 = \sqrt{\frac{1}{1.12^2 \pi a} K_{Ic}}$$

$$\vec{T}^d = \sigma_0 \vec{e}_z \quad \text{far away from crack}$$

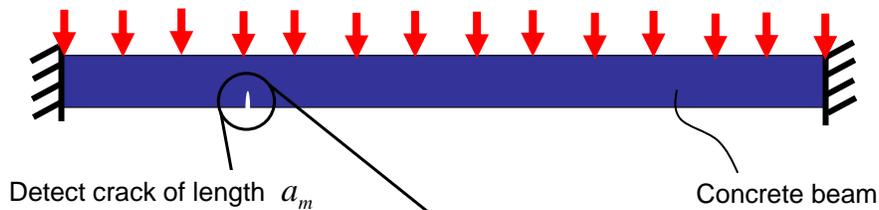


"Surface crack"

$$\vec{T}^d = -\sigma_0 \vec{e}_z$$

23

# Example application



Photograph of crack (crack length of a(sub m) removed due to copyright restrictions.



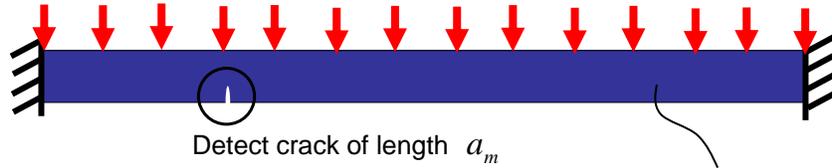
Measure length of crack  $a_m$

E.g. add fluorescent fluid use UV light

24

[http://www.amesresearch.com/images/cshst/block\\_before.jpg](http://www.amesresearch.com/images/cshst/block_before.jpg)

## Example application



**Question:** Will structure fail?

Concrete beam

**Solution:** Solving beam problem provides us with stress distribution in section

$$\sigma_{xx}(z; x) = \frac{M_y(x)}{I} z \quad \sigma_{xx}(z) = \frac{\sigma_0}{h/2} z$$

The diagram shows a vertical coordinate  $z$  with the top at  $h/2$  and the bottom at  $-h/2$ . A red line represents the linear stress distribution  $\sigma_{xx}(z)$ , starting at zero at the neutral axis and increasing linearly to its maximum value at the top and minimum at the bottom.

Calculate critical fracture stress and compare with stress in beam structure

The diagram shows a cross-section of a beam with a vertical crack. The crack length is indicated as  $a_m$ .

$$\sigma_{0,crit} = \sqrt{\frac{1}{1.12^2 \pi a_m}} K_{Ic} \begin{cases} \sigma_{xx}(z = -\frac{h}{2}) \geq \sigma_{0,crit} & \text{failure} \\ \sigma_{xx}(z = -\frac{h}{2}) < \sigma_{0,crit} & \text{no failure} \end{cases}$$