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16.346 Astrodynamics
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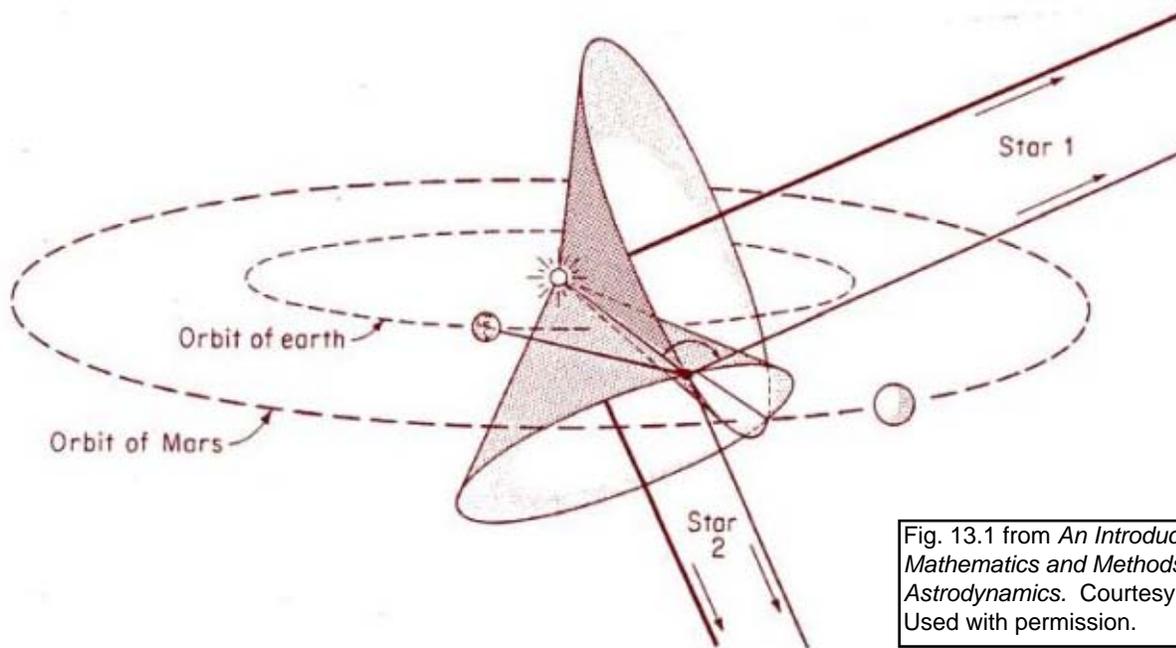


Fig. 13.1 from *An Introduction to the Mathematics and Methods of Astrodynamics*. Courtesy of AIAA. Used with permission.

The Line of Position

$$\begin{aligned} \mathbf{i}_r \cdot \mathbf{i}_{*1} &= -\cos A_1 \\ \mathbf{i}_r \cdot \mathbf{i}_{*2} &= -\cos A_2 \end{aligned} \quad \Rightarrow \quad \mathbf{i}_r = \alpha \mathbf{i}_{*1} + \beta \mathbf{i}_{*2} + \gamma \mathbf{i}_{*1} \times \mathbf{i}_{*2}$$

where

$$\alpha \sin^2 \varphi = \cos A_2 \cos \varphi - \cos A_1$$

$$\beta \sin^2 \varphi = \cos A_1 \cos \varphi - \cos A_2$$

$$\gamma^2 \sin^2 \varphi = 1 + \alpha \cos A_1 + \beta \cos A_2$$

and $\cos \varphi = \mathbf{i}_{*1} \cdot \mathbf{i}_{*2}$

The Position Fix

$$\mathbf{i}_r \cdot \mathbf{i}_{*1} = -\cos A_1$$

$$\mathbf{i}_r \cdot \mathbf{i}_{*2} = -\cos A_2$$

$$\mathbf{i}_r \cdot \mathbf{r}_p = r - |\mathbf{r}_p - \mathbf{r}| \cos A_3$$

where \mathbf{r}_p is the position vector of a planet or other near object.

Henceforth, we will linearize the measurements so that we can deal with a set of redundant measurements using Gauss's Method of Least Squares.

Determining the Measurement Geometry Vector

#13.2

For an arbitrary angle $A(\mathbf{r})$, we calculate the measurement geometry vector \mathbf{h} from the Taylor Series expansion about a reference position \mathbf{r}_0 and discarding all terms of higher order in δr :

$$A(\mathbf{r}) = A(\mathbf{r}_0) + \left. \frac{\partial A}{\partial \mathbf{r}} \right|_{\mathbf{r}=\mathbf{r}_0} \delta \mathbf{r} + \dots = A_0 + \mathbf{h}^T \delta \mathbf{r} + \dots$$

Hence

$$\delta A = \mathbf{h}^T \delta \mathbf{r} \quad \text{where} \quad \mathbf{h}^T = \left. \frac{\partial A}{\partial \mathbf{r}} \right|_{\mathbf{r}=\mathbf{r}_0}$$

Measuring the Angle between a Near Object and a Star

The angle between the line-of-sight to a near object, e.g., the sun or a planet, and the line-of-sight to a distant star is defined by

$$r \cos A = -\mathbf{i}_*^T \mathbf{r}$$

from which

$$\frac{\partial r}{\partial \mathbf{r}} \cos A - r \sin A \frac{\partial A}{\partial \mathbf{r}} = -\mathbf{i}_*^T$$

The derivative of the scalar r with respect to the vector \mathbf{r} is obtained from

$$r^2 = \mathbf{r} \cdot \mathbf{r} = \mathbf{r}^T \mathbf{r} \quad \implies \quad 2r \frac{\partial r}{\partial \mathbf{r}} = 2\mathbf{r}^T \frac{\partial \mathbf{r}}{\partial \mathbf{r}} = 2\mathbf{r}^T \mathbf{I} \quad \implies \quad \frac{\partial r}{\partial \mathbf{r}} = \frac{1}{r} \mathbf{r}^T = \mathbf{i}_r^T$$

Therefore,

$$\mathbf{h} = \frac{1}{r \sin A} (\cos A \mathbf{i}_r + \mathbf{i}_*) \quad \text{or} \quad \mathbf{h} = \frac{1}{r} \mathbf{i}_n$$

The vector \mathbf{i}_n is a unit vector in the plane of the measurement and perpendicular to the line-of-sight to the near object.

Measuring the Angle between Two Near Objects

The angle between the two position vectors \mathbf{r} and \mathbf{d} produces the measurement equation

$$\mathbf{d}^T \mathbf{r} = d r \cos A$$

Since $\mathbf{r} - \mathbf{d} = \text{constant}$, then $\delta \mathbf{d} = \delta \mathbf{r}$. Again, from $d^2 = \mathbf{d}^T \mathbf{d}$, we have

$$2d \frac{\partial d}{\partial \mathbf{r}} = 2\mathbf{d}^T \mathbf{I} \quad \text{or} \quad \frac{\partial d}{\partial \mathbf{r}} = \mathbf{i}_d^T$$

Hence:

$$\mathbf{h} = -\frac{1}{r \sin A} (\mathbf{i}_d - \cos A \mathbf{i}_r) - \frac{1}{d \sin A} (\mathbf{i}_r - \cos A \mathbf{i}_d) \quad \text{or} \quad \mathbf{h} = \frac{1}{r} \mathbf{i}_n + \frac{1}{d} \mathbf{i}_m$$

Both \mathbf{i}_n and \mathbf{i}_m are unit vectors in the plane determined by the spacecraft and the two near bodies.

The Measurement Geometry Matrix

For several measurements we define

$$\mathbf{H} = [\mathbf{h}_1 \quad \mathbf{h}_2 \quad \cdots \quad \mathbf{h}_n] \quad \text{so that} \quad \delta \mathbf{q} = \mathbf{H}^T \delta \mathbf{r}$$

where

$$\delta \mathbf{q} = \begin{bmatrix} \delta q_1 \\ \delta q_2 \\ \vdots \\ \delta q_n \end{bmatrix} \quad \text{and} \quad \delta \mathbf{r} = \begin{bmatrix} \delta x \\ \delta y \\ \delta z \end{bmatrix}$$

Gauss' Method of Least Squares

#13.5

Given m_{ij} and c_i : To determine x_i so that

$$\sum_{j=1}^n m_{ij} x_j = c_i \quad \text{where} \quad i = 1, 2, \dots, N > n$$

is "as nearly satisfied as possible."

- Define: **Residuals** $e_i = \sum_{j=1}^n m_{ij} x_j - c_i$
- Choose: **Weighting factor** $w_i > 0$ for i^{th} residual
- Determine: x_1, x_2, \dots, x_n so that $w_1 e_1^2 + w_2 e_2^2 + \cdots + w_N e_N^2$ is a minimum.

Solution of Least Squares Problem

- Vector of residuals: $\mathbf{e} = \mathbf{M}\mathbf{x} - \mathbf{c}$
- Weighting matrix: $\mathbf{W} = \begin{bmatrix} w_1 & 0 & \cdots & 0 \\ 0 & w_2 & \cdots & 0 \\ \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & \cdots & w_N \end{bmatrix} = \mathbf{W}^T$
- Weighted squares:

$$\mathbf{e}^T \mathbf{W} \mathbf{e} = (\mathbf{x}^T \mathbf{M}^T - \mathbf{c}^T) \mathbf{W} (\mathbf{M}\mathbf{x} - \mathbf{c}) = \mathbf{x}^T \mathbf{M}^T \mathbf{W} \mathbf{M} \mathbf{x} - \mathbf{c}^T \mathbf{W} \mathbf{M} \mathbf{x} - \mathbf{x}^T \mathbf{M}^T \mathbf{W} \mathbf{c} + \mathbf{c}^T \mathbf{W} \mathbf{c}$$

- Least value of the weighted squares:

$$\frac{\partial}{\partial \mathbf{x}} (\mathbf{e}^T \mathbf{W} \mathbf{e}) = 2\mathbf{x}^T \mathbf{M}^T \mathbf{W} \mathbf{M} - 2\mathbf{c}^T \mathbf{W} \mathbf{M} = \mathbf{0}^T \quad \text{or} \quad \mathbf{M}^T \mathbf{W} \mathbf{M} \mathbf{x} = \mathbf{M}^T \mathbf{W} \mathbf{c}$$

$$\boxed{\mathbf{x} = (\mathbf{M}^T \mathbf{W} \mathbf{M})^{-1} \mathbf{M}^T \mathbf{W} \mathbf{c}}$$

Note: If $y = \mathbf{x}^T \mathbf{B} \mathbf{x} = (\mathbf{x}^T \mathbf{B} \mathbf{x})^T = \mathbf{x}^T \mathbf{B}^T \mathbf{x}$ then $\frac{\partial y}{\partial \mathbf{x}} = \mathbf{x}^T \mathbf{B} \mathbf{I} + \mathbf{x}^T \mathbf{B}^T \mathbf{I}$

Application of Gauss' Method of Least Squares to Space Navigation

$$\mathbf{x} = (\mathbf{M}^T \mathbf{W} \mathbf{M})^{-1} \mathbf{M}^T \mathbf{W} \mathbf{c}$$

In our notation

$$\mathbf{x} = \delta \hat{\mathbf{r}} \quad \mathbf{M}^T = \mathbf{H} \quad \mathbf{W} = \mathbf{A}^{-1} \quad w_i = \frac{1}{\sigma_i^2} \quad \mathbf{c} = \delta \tilde{\mathbf{q}}$$

so that

$$\delta \hat{\mathbf{r}} = \mathbf{F} \delta \tilde{\mathbf{q}}$$

\mathbf{F} is called the Estimator Matrix where

$$\mathbf{F} = \mathbf{P} \mathbf{H} \mathbf{A}^{-1} \quad \text{and} \quad \mathbf{P} = (\mathbf{H} \mathbf{A}^{-1} \mathbf{H}^T)^{-1}$$

The deviation from the reference position is an estimate, denoted by the “hat” over the position vector

$$\delta \hat{\mathbf{r}} = \delta \mathbf{r} + \boldsymbol{\epsilon}$$

and is the sum of the actual deviation and the error in the estimate. Similarly,

$$\delta \tilde{\mathbf{q}} = \delta \mathbf{q} + \boldsymbol{\alpha}$$

where $\boldsymbol{\alpha}$ is the vector error in the determination of the quantities measured. (The vector $\delta \mathbf{q}$ is the actual deviation in those quantities from their reference values.)

The Information Matrix

The matrix \mathbf{P}^{-1} is called the Information Matrix because of the property

$$\mathbf{H} \mathbf{A}^{-1} \mathbf{H}^T = \frac{\mathbf{h}_1 \mathbf{h}_1^T}{\sigma_1^2} + \frac{\mathbf{h}_2 \mathbf{h}_2^T}{\sigma_2^2} + \dots$$

Each new measurement adds a new term to the series and each term contains all the information about the new measurement.

$$\mathbf{P}^{-1} = \mathbf{H} \mathbf{A}^{-1} \mathbf{H}^T = \sum_{i=1}^N \frac{\mathbf{h}_i \mathbf{h}_i^T}{\sigma_i^2}$$

The factor σ_i^2 is called the variance. The larger the i^{th} variance the less weight is given to the i^{th} measurement.